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SATELLITE ORBITAL DATA

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SATELLITE ORBITAL DATA

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NOTE

It is in the nature of frequently used computer programs that from time to time they undergo modifications as dictated by the experience of people who use them. Thus our Differential Orbit Improvement Program, being in use since the end of 1958, has been changed on several occasions. In the past we did not give notice of any of these changes, because they affected merely the internal structure and the capabilities of the program. Some recent modifications, however, should be pointed out because they alter the definition of the mean orbital elements.

1. As before, the semimajor axis a of an orbit is being computed from the mean motion n of the satellite according to the formula

$$a = \sqrt[3]{\frac{GM}{n^2}} \left\{ 1 - \frac{J_2 a_E^2}{2p^2} \sqrt{1 - e^2} \left(1 - \frac{3}{2} \sin^2 I \right) \right\}$$

(Y. Kozai, Astron. Journ., vol. 64, pp. 367-377; in his equation (14) we put $A_2 = \frac{3}{2} J_2 a_E^2$). In the old program, the mean motion in turn was defined as the time derivative of the mean anomaly M . Therefore in cases where in addition to a polynomial part the mean anomaly also had a trigonometric part, the program produced small but unwanted, long-periodic variations in the semimajor axis. In the new program the mean motion is defined as the time derivative of the mean anomaly's polynomial part only.

2. The old program provided internally only for those first-order short-periodic perturbations that are caused by the second zonal harmonic (J_2 -term) of the geopotential. The new program has the optional capability to account for lunar perturbations with periods of approximately two weeks. Their analytical expressions are quite complicated and will not be given here. As a rule, we use this feature of the program only in connection with orbits that are computed from precisely reduced Baker-Nunn observations. In our future publications of satellite orbital data we will always mention if lunar perturbations were included in the computations.

A detailed write-up of the new Differential Orbit Improvement Program, henceforth called DOI-3, will be given shortly by Mr. Edward M. Gaposchkin.

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ORBITAL INFORMATION¹

The orbital elements have been derived by the indicated staff members of the Satellite Tracking Program, Smithsonian Astrophysical Observatory, employing the SAO Differential Orbit Improvement Program (DOI).

Field-reduced photographs from SAO Baker-Nunn cameras comprise the majority of observations used in computing these orbital data. SAO Moonwatch teams, the NASA Minitrack network, foreign observatories, miscellaneous U.S. and foreign observers, and various radar installations also contribute valuable observations.

As opposed to osculating elements, the elements presented here are mean elements in the sense that the effects of the short-period perturbations due to the earth's oblateness have been eliminated.

SAO mean elements have been derived from observations covering several days and are given in the form of a table. The successive sets of elements are essentially independent of each other. They are dependent, however, in the sense that high-order coefficients in the secular and the long-periodic terms are generally considered as known and as constant for periods of several weeks or months, as dictated by convenience.

The times of epoch in the mean elements are reckoned in Julian Days, but for the sake of convenience the number 2400000.5 has been subtracted to provide an abbreviated notation which we call "Modified Julian Days," or "MJD."

The units of the orbital elements are degrees for angular quantities, megameters ($M_m = 10^6$ meters) for linear quantities, and revolutions for the mean anomaly M and its derivatives.

The tabulated values of the SAO mean elements give the values of argument of perigee ω , right ascension of the ascending node Ω , inclination i , eccentricity e , and mean anomaly M as functions of time $t = T - T_0$ (where T_0 is the reference epoch) expressed in days. The single digit placed at the right of each value represents the standard error for that element and refers to the last digit given.

The same tabulation also gives the mean (anomalistic) motion n , the orbital acceleration $n'/2$ or $n'(dn/dt)$, and the semimajor axis a or the geocentric distance of perigee q (in megameters). Of the last three columns, the one headed N indicates the number of observations used for the computation of a set of elements; the one headed D , the number of days used; and the one headed σ , the standard error of the representation of the observations relative to their assumed accuracy.

SAO smoothed elements have been derived from observations covering about two weeks or more. They are given as functions of time and generally include both secular and periodic terms. The general expression for any element E is

¹This work was supported in part by grant NsG 87-60 from the National Aeronautics and Space Administration.

$$E = E_0 + E_1 t + E_2 t^2 + \dots + \sum A_i \sin(B_i + C_i t)$$

where $t = T - T_0$ is again expressed in days. The presence of a standard error associated with a particular coefficient indicates that this quantity was determined by the process of differential orbit improvement; the absence of a standard error means that the quantity was taken from some other source.

In our computer program, the inclination and the argument of perigee are referred to the true equator of date; the right ascension of the ascending node, however, is reckoned from the mean equinox of 1950.0 along the corresponding mean equator to the intersection with the moving true equator of date, and then along the true equator of date. To transform from right ascension of the node as determined by the D ϕ I to right ascension of the node referred to the mean equinox of date, one uses

$$\Omega^\circ = \Omega^\circ(D\phi I) + 3^\circ 508 \times 10^{-5} (MJD - 33281) ,$$

where MJD stands for the Modified Julian Day of the date.

The mean (anomalistic) motion n can be obtained from the smoothed elements by differentiating the expression for M , and the orbital acceleration n' can be obtained by twice differentiating the same expression for M .

The sun-perigee data are related to the perturbing effects of atmospheric drag. From left to right are the Modified Julian Day (MJD); the perigee height Z (in kilometers) above the International Ellipsoid of Reference; the geocentric latitude of the perigee (φ); the angular geocentric distance (ψ) from the perigee of the sun; and the difference in right ascension (D.R.A.) between the perigee and the sun; all these angles are expressed in degrees. In the last column we give the rate of change of the period (\dot{P}) expressed in days per day.

Satellite 1958 Alpha 1 (Explorer 1)

Beatrice Miller

I. SAO smoothed elements

The following elements are based on 31 observations and are valid for the period April 1 through April 15, 1963.

$$T_0 = 38128.0 \text{ MJD}$$

$$\omega = (268^\circ 06 \pm 3) + (7^\circ 600 \pm 6)t - .000296t^2 + .3144 \cos \omega$$

$$\Omega = (322^\circ 510 \pm 7) - (5^\circ 086 \pm 2)t - .36 \times 10^{-4}t^2 + .0031 \cos \omega$$

$$i = (33^\circ 203 \pm 2) - .000476t + .129 \times 10^{-4}t^2 - .0039 \sin \omega$$

$$e = (.08981 \pm 2) + .5249 \times 10^{-4}t - .1626 \times 10^{-5}t^2 + .0004980 \sin \omega$$

$$M = (.72097 \pm 8) + (13.68489 \pm 2)t + (.64 \pm 1) \times 10^{-4}t^2 \\ - (.26 \pm 11) \times 10^{-6}t^3 - (.32 \pm 19) \times 10^{-7}t^4 - .0008998 \cos \omega$$

Standard error of one observation: $\sigma = \pm 2.88$.

The following elements are based on 54 observations and are valid for the period April 15 through May 1, 1963.

$$T_0 = 38142.0 \text{ MJD}$$

$$\omega = (14^\circ 43 \pm 1) + (7^\circ 599 \pm 2)t - .000296t^2 + .3144 \cos \omega$$

$$\Omega = (251^\circ 279 \pm 4) - (5^\circ 0884 \pm 6)t - .36 \times 10^{-4}t^2 + .0031 \cos \omega$$

$$i = (33^\circ 201 \pm 1) - .000114t + .129 \times 10^{-4}t^2 - .0039 \sin \omega$$

$$e = (.08944 \pm 2) + .697 \times 10^{-5}t - .1626 \times 10^{-5}t^2 + .0004980 \sin \omega$$

$$M = (.32221 \pm 4) + (13.686831 \pm 7)t + (.701 \pm 5) \times 10^{-4}t^2 \\ - (.77 \pm 4) \times 10^{-6}t^3 - (.21 \pm 7) \times 10^{-7}t^4 - .0008998 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.53$.

The following elements are based on 75 observations and are valid for the period May 1 through May 16, 1963.

$$T_0 = 38158.0 \text{ MJD}$$

$$\omega = (136^\circ 00 \pm 1) + (7^\circ 603 \pm 2)t - 0.000296t^2 + 0.3144 \cos \omega$$

$$\Omega = (169^\circ 0866 \pm 4) - (5^\circ 0886 \pm 8)t - 0.36 \times 10^{-4}t^2 + 0.0031 \cos \omega$$

$$i = (33^\circ 203 \pm 1) + 0.000298t + 0.129 \times 10^{-4}t^2 - 0.0039 \sin \omega$$

$$e = (.08929 \pm 2) - .4507 \times 10^{-4}t - .1626 \times 10^{-5}t^2 + .0004980 \sin \omega$$

$$M = (.32865 \pm 2) + (13.689148 \pm 5)t + (.933 \pm 6) \times 10^{-4}t^2 \\ + (.95 \pm 4) \times 10^{-6}t^3 - (.19 \pm 10) \times 10^{-7}t^4 - .0008998 \cos \omega$$

Standard error of one observation: $\sigma = \pm 2.00$.

The following elements are based on 124 observations and are valid for the period May 16 through June 1, 1963.

$$T_0 = 38174.0 \text{ MJD}$$

$$\omega = (257^\circ 80 \pm 3) + (7^\circ 531 \pm 2)t - 0.001255t^2 + 0.3144 \cos \omega$$

$$\Omega = (88^\circ 420 \pm 3) - (5^\circ 0928 \pm 7)t - 0.000121t^2 + 0.0031 \cos \omega$$

$$i = (33^\circ 2016 \pm 8) - 0.000199t - 0.11 \times 10^{-5}t^2 - 0.0039 \sin \omega$$

$$e = (.08896 \pm 4) + .000158t - .38 \times 10^{-5}t^2 + .0004980 \sin \omega$$

$$M = (.37731 \pm 9) + (13.691874 \pm 8)t + (.447 \pm 4) \times 10^{-4}t^2 \\ - (.11 \pm 4) \times 10^{-6}t^3 + (.13 \pm 8) \times 10^{-6}t^4 - .0008998 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.60$.

The following elements are based on 38 observations and are valid for the period June 1 through June 15, 1963.

$$T_0 = 38188.0 \text{ MJD}$$

$$\omega = (4^\circ 03 \pm 2) + (7^\circ 584 \pm 4)t - 0.001255t^2 + 0.3144 \cos \omega$$

$$\Omega = (17^\circ 141 \pm 7) - (5^\circ 091 \pm 2)t - 0.000121t^2 + 0.0031 \cos \omega$$

$$i = (33^\circ 205 \pm 3) - 0.000230t - 0.11 \times 10^{-5}t^2 - 0.0039 \sin \omega$$

$$e = (.08926 \pm 4) + .52 \times 10^{-4}t - .38 \times 10^{-5}t^2 + .0004980 \sin \omega$$

$$M = (.07238 \pm 5) + (13.69361 \pm 1)t + (.111 \pm 1) \times 10^{-3}t^2 \\ + (.208 \pm 8) \times 10^{-5}t^3 - (.17 \pm 2) \times 10^{-6}t^4 - .0008998 \cos \omega$$

Standard error of one observation: $\sigma = \pm 3.03$.

The following elements are based on 109 observations and are valid for the period June 15 through July 1, 1963.

$$T_0 = 38204.0 \text{ MJD}$$

$$\omega = (125^\circ 71 \pm 2) + (7^\circ 628 \pm 5)t - 0^\circ 001255t^2 + 0^\circ 3144 \cos \omega$$

$$\Omega = (295^\circ 627 \pm 5) - (5^\circ 0969 \pm 7)t - 0^\circ 000121t^2 + 0^\circ 0031 \cos \omega$$

$$i = (33^\circ 202 \pm 2) - 0^\circ 000265t - 0^\circ 11 \times 10^{-5}t^2 - 0^\circ 0039 \sin \omega$$

$$e = (.08909 \pm 2) - .70 \times 10^{-4}t - .38 \times 10^{-5}t^2 + .0004980 \sin \omega$$

$$M = (.19902 \pm 7) + (13.69712 \pm 1)t + (.908 \pm 6) \times 10^{-4}t^2$$

$$- (.68 \pm 4) \times 10^{-6}t^3 + (.154 \pm 9) \times 10^{-6}t^4 - .0008998 \cos \omega$$

Standard error of one observation: $\sigma = \pm 2^\circ 95$.

T (MTD)	ω	Ω	i	e	M	n	n'/2	q	N	D	σ
38120.0	207.09 4	3.180 9	33.205 2	.00d943 2	.24647 9	13.683877 2	.62E-4 1	6.721730	49	8	1.80
38124.0	237.51 2	342.860 8	33.208 2	.08920 2	.98288 5	13.684373 2	.63E-4 8	6.723296	23	8	.99
38128.0	268.04 3	322.51 1	33.208 3	.08917 3	.72108 7	13.684889 4	.61E-4 1	6.723311	13	8	1.00
38132.0	298.60 3	302.156 7	33.202 3	.08917 5	.4610 1	13.685351 7	.71E-4 2	6.723192	12	8	.67
38136.0	329.08 4	281.796 9	33.196 4	.0894 2	.2033 2	13.685952 3	.76E-4 1	6.721e-1 8	21	8	.51
38140.0	359.52 4	261.457 7	33.199 2	.0895 2	.9480 2	13.686557 2	.739E-4 8	.e-120684	29	8	.55
38144.0	29.90 2	241.108 7	33.198 2	.08965 3	.69535 4	13.687118 2	.64E-4 1	6.719091	28	8	.68
38148.0	60.18 1	220.747 4	33.197 9	.08991 2	.44504 2	13.687597 1	.616E-4 5	6.717027	26	8	.43
38152.0	90.378 6	200.401 3	33.197 6	.08993 2	.19686 1	13.6881796 9	.854E-4 6	6.716652	34	8	.49
38156.0	120.56 1	180.042 5	33.199 1	.08974 2	.95137 2	13.688847 1	.836E-4 6	6.717860	22	8	.58
38160.0	150.92 1	159.689 4	33.201 1	.08960 3	.70812 3	13.689599 2	.1070E-3 8	6.718667	33	8	.63
38164.0	181.19 4	139.332 5	33.200 2	.0895 1	.4685 2	13.690414 2	.870E-4 8	6.718982	76	8	.63
38168.0	211.66 4	118.963 4	33.2026 8	.08901 8	.2310 1	13.691027 1	.693E-4 5	6.722533	83	8	.54
38172.0	242.29 1	98.593 6	33.207 1	.08884 3	.99527 3	13.691504 2	.467E-4 8	6.723601	68	8	.72
38176.0	272.77 1	78.230 6	33.208 2	.08856 3	.76147 2	13.691693 2	.62E-4 1	6.725653	63	8	.88
38180.0	303.34 1	57.881 7	33.205 2	.08880 3	.52904 3	13.692214 2	.77E-4 1	6.723660	51	8	.94
38184.0	333.94 2	37.51 1	33.206 3	.08902 4	.29878 5	13.692842 3	.78E-4 1	6.721876	25	8	1.22
38188.0	4.37 3	17.14 1	33.200 4	.08927 6	.07142 8	13.693568 4	.105E-3 2	6.719753	20	8	1.25
38192.0	34.67 1	356.774 3	33.202 2	.08942 3	.84776 4	13.694494 2	.1137E-3 8	6.718338	22	8	.50
38196.0	64.99 2	336.384 5	33.197 3	.08949 2	.62776 7	13.695452 2	.128E-3 1	6.717503	36	8	.66
38200.0	95.25 3	315.999 7	33.197 3	.08946 3	.41188 9	13.696491 3	.123E-3 1	6.717392	43	8	1.04
38204.0	125.63 1	295.623 3	33.198 1	.08945 1	.19926 4	13.697294 1	.840E-4 5	6.717236	40	8	.49
38208.0	156.85 2	275.234 3	33.201 1	.08911 1	.98996 5	13.698035 1	.1076E-3 6	6.719466	59	8	.66

Table 1

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1958 ALPHA

MJD	Z	Ψ	ψ	D.R.A.	P
PERIGEE IN EARTH SHADOW					
38126.	345.	-14.4	160.7	196.6	-0.662E-06
38124.	349.	-27.5	149.6	202.2	-0.673E-06
38128.	351.	-33.2	139.8	213.2	-0.651E-06
38132.	350.	-28.7	133.7	224.6	-0.758E-06
38136.	345.	-16.3	130.3	230.8	-0.812E-06
38140.	342.	-0.3	126.2	233.0	-0.789E-06
38144.	342.	15.8	118.5	235.0	-0.683E-06
PERIGEE IN SUNLIGHT					
38148.	343.	28.4	107.5	240.7	-0.658E-06
38152.	345.	33.2	96.4	251.4	-0.912E-06
38156.	344.	28.1	89.1	262.0	-0.892E-06
38160.	342.	15.4	87.6	267.6	-0.114E-05
38164.	341.	-0.7	90.9	269.2	-0.928E-06
38168.	346.	-16.7	94.4	271.2	-0.739E-06
38172.	350.	-29.0	93.6	277.4	-0.498E-06
38176.	354.	-33.2	87.1	288.5	-0.661E-06
38180.	350.	-27.2	76.7	298.9	-0.821E-06
38184.	345.	-13.9	65.7	304.0	-0.832E-06
38188.	341.	2.4	56.6	305.5	-0.112E-05
38192.	342.	18.1	49.3	307.3	-0.121E-05
38196.	344.	29.7	41.7	313.6	-0.136E-05
38200.	345.	33.0	32.6	324.5	-0.131E-05
38204.	343.	26.4	23.5	334.3	-0.895E-06
38208.	342.	12.4	22.4	339.4	-0.115E-05

Satellite 1959 Alpha 1 (Vanguard 2)

Maria Gutierrez

I. SAO smoothed elements

The following elements are based on 91 observations and are valid for the period April 1 through May 1, 1963.

$$T_0 = 38134.0 \text{ MJD}$$

$$\omega = (232^\circ 390 \pm 4) + (5^\circ 2919 \pm 4)t - 0.59 \times 10^{-4}t^2 + 0.1523 \cos \omega$$

$$\Omega = (249^\circ 817 \pm 2) - (3^\circ 5209 \pm 2)t - 0.42 \times 10^{-5}t^2 + 0.0077 \cos \omega$$

$$i = (32^\circ 8772 \pm 5) - 0.0069 \sin \omega$$

$$e = (.16431 \pm 1) - .138 \times 10^{-5}t + .000457 \sin \omega$$

$$M = (.638852 \pm 7) + (11.4792753 \pm 8)t + (.395 \pm 4) \times 10^{-5}t^2 \\ - (.22 \pm 3) \times 10^{-7}t^3 - .000439 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.00$.

The following elements are based on 126 observations and are valid for the period May 1 through June 1, 1963.

$$T_0 = 38164.0 \text{ MJD}$$

$$\omega = (31^\circ 121 \pm 5) + (5^\circ 2957 \pm 4)t + 0.000114t^2 + 0.1523 \cos \omega$$

$$\Omega = (144^\circ 206 \pm 2) - (3^\circ 5206 \pm 2)t + 0.11 \times 10^{-5}t^2 + 0.0077 \cos \omega$$

$$i = (32^\circ 8790 \pm 7) - 0.0069 \sin \omega$$

$$e = (.16433 \pm 1) - .75 \times 10^{-5}t + .000457 \sin \omega$$

$$M = (.01993 \pm 1) + (11.4794423 \pm 7)t + (.243 \pm 2) \times 10^{-5}t^2 \\ - (.77 \pm 26) \times 10^{-8}t^3 - .000439 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.33$.

The following elements are based on 109 observations and are valid for the period June 1 through July 1, 1963.

$$T_0 = 38184.0 \text{ MJD}$$

$$\omega = (136^\circ 941 \pm 7) + (5^\circ 2933 \pm 5)t + 0.000114t^2 + 0.1523 \cos \omega$$

$$\Omega = (73^\circ 796 \pm 4) - (3^\circ 5202 \pm 2)t + 0.11 \times 10^{-5}t^2 + 0.0077 \cos \omega$$

$$i = (32^\circ 8795 \pm 9) - 0.0069 \sin \omega$$

$$e = (.164384 \pm 8) - .75 \times 10^{-5}t + .000457 \sin \omega$$

$$M = (.61000 \pm 2) + (11.479543 \pm 2)t + (.21 \pm 1) \times 10^{-5}t^2 \\ - (.13 \pm 3) \times 10^{-7}t^3 - .000439 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.35$.

II. SAO mean elements -- Satellite 1959 Alpha 1

3 April - 30 June 1963

T (MD)	w	Ω	i	e	M	n	$n^{1/2}$	q	N	D	σ
38122.0	168.722 9	292.057 6	32.877 2	.16439 3	.88862 3	11.479172 2	.5E-5 1	6.935173	23	8	.37
38126.0	189.911 7	277.974 4	32.879 2	.16422 3	.80533 2	11.479207 2	.53E-5 6	6.936560	16	8	.34
38130.0	211.10 1	263.892 7	32.878 3	.16406 3	.72218 2	11.479242 1	.33E-5 6	6.937887	11	8	.43
38134.0	232.289 7	249.813 5	32.881 1	.16398 2	.63913 1	11.479273 1	.40E-5 3	6.938525	15	8	.34
38138.0	253.510 5	235.734 4	32.883 8	.16392 2	.556139 8	11.479305 1	.41E-5 2	6.939063	28	8	.35
38142.0	274.737 8	221.652 7	32.884 1	.16385 2	.47325 1	11.479334 1	.31E-5 7	6.939574	34	8	.55
38146.0	295.944 5	207.565 3	32.8863 8	.16390 2	.39052 1	11.479357 9	.29E-5 4	6.939172	31	8	.36
38150.0	317.154 7	193.489 4	32.883 1	.16397 2	.30785 1	11.479399 1	.8E-6 5	6.938545	37	8	.46
38154.0	338.337 9	179.413 5	32.880 1	.16411 2	.22533 2	11.479421 1	.41E-5 8	6.937353	36	8	.45
38158.0	359.51 1	165.324 7	32.879 2	.16423 3	.14297 2	11.479449 2	.33E-5 6	6.936331	29	8	.38
38162.0	2C.70 2	151.262 5	32.870 2	.16454 5	.06055 5	11.479469 2	.2E-5 1	6.933754	21	8	.46
38166.0	41.87 1	137.172 3	32.873 2	.16459 2	.97840 2	11.479482 3	.1E-6 9	6.933329	38	8	.60
38170.0	62.979 7	123.087 3	32.873 1	.16470 2	.89645 2	11.479493 7	.28E-5 4	6.932429	47	8	.54
38174.0	84.086 8	108.999 4	32.872 1	.16477 2	.81460 2	11.479512 1	.18E-5 4	6.931876	27	8	.42
38178.0	105.18 1	94.922 5	32.872 1	.16478 2	.73283 2	11.479528 1	.22E-5 6	6.931735	26	8	.56
38182.0	126.30 1	80.836 4	32.874 1	.16469 2	.65110 2	11.479546 1	.21E-5 6	6.932528	35	8	.61
38186.0	147.42 1	66.744 5	32.877 1	.16456 2	.56942 2	11.479559 1	.24E-5 5	6.933563	25	8	.56
38190.0	168.568 8	52.670 4	32.878 2	.16438 1	.48772 2	11.479575 1	.15E-5 4	6.935040	21	8	.40
38194.0	189.737 9	38.591 4	32.880 2	.16420 1	.40603 2	11.479590 1	.23E-5 6	6.936536	27	8	.45
38198.0	210.93 1	24.512 5	32.880 3	.16404 2	.32434 3	11.479605 1	.17E-5 6	6.937884	28	8	.48
38202.0	232.157 9	10.420 4	32.887 3	.16393 1	.24266 2	11.479614 1	.10E-5 5	6.938833	38	8	.51
38206.0	253.385 9	356.344 3	32.883 2	.16383 1	.16098 2	11.479625 1	.13E-5 4	6.939583	35	8	.47
38210.0	274.59 1	342.265 6	32.887 3	.16384 2	.07942 3	11.479634 1	.2E-7 9	6.939513	21	8	.60

Table 2

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1959 ALPHA 1

MJD	Z	φ	ψ	D. R.A.	\dot{P}
PERIGEE IN SUNLIGHT					
38122.	557.	6.1	90.5	91.0	-0.759E-07
38126.	558.	-5.4	91.7	91.1	-0.804E-07
38130.	561.	-16.3	94.1	91.9	-0.501E-07
38134.	564.	-25.4	98.2	94.7	-0.607E-07
38138.	566.	-31.4	104.2	100.1	-0.622E-07
38142.	567.	-32.8	111.1	107.3	-0.470E-07
PERIGEE IN EARTH SHADOW					
38146.	566.	-29.2	117.3	113.9	-0.440E-07
38150.	563.	-21.7	121.1	118.1	-0.121E-07
38154.	560.	-11.6	121.4	119.6	-0.622E-07
38158.	558.	-0.3	118.4	119.7	-0.501E-07
38162.	556.	11.1	113.8	119.8	-0.304E-07
PERIGEE IN SUNLIGHT					
38166.	558.	21.2	109.6	121.1	0.152E-08
38170.	559.	28.9	107.7	124.8	-0.425E-07
38174.	560.	32.7	108.9	130.9	-0.273E-07
PERIGEE IN EARTH SHADOW					
38178.	559.	31.6	113.3	137.7	-0.334E-07
38182.	558.	25.9	120.0	142.8	-0.319E-07
38186.	557.	17.0	127.8	145.2	-0.364E-07
38190.	557.	6.2	135.5	145.6	-0.228E-07
38194.	558.	-5.3	142.0	145.2	-0.349E-07
38198.	561.	-16.2	146.8	145.5	-0.258E-07
38202.	564.	-25.4	150.7	147.8	-0.152E-07
38206.	567.	-31.3	154.6	152.8	-0.197E-07
38210.	567.	-32.8	159.6	159.5	0.304E-09

I. SAO smoothed elements

The following elements are based on 103 observations and are valid for the period April 1 through May 1, 1963.

$$T_0 = 38136.0 \text{ MJD}$$

$$\omega = (39^\circ 264 \pm 4) + (4^\circ 8904 \pm 4)t + .000134t^2 + .1295 \cos \omega$$

$$\Omega = (263^\circ 9848 \pm 2) - (3^\circ 28803 \pm 2)t + (.10 \times 10^{-4})t^2 + .0090 \cos \omega$$

$$i = (33^\circ 353 \pm 1) + .000185t - .0077 \sin \omega$$

$$e = (.18879 \pm 2) - (.62 \times 10^{-5})t + .000452 \sin \omega$$

$$M = (.830364 \pm 8) + (11.088466 \pm 1)t - (.54 \pm 3) \times 10^{-6}t^2 \\ + (.11 \pm 3) \times 10^{-7}t^3 - .000376 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.35$.

The following elements are based on 283 observations and are valid for the period May 1 through June 1, 1963.

$$T_0 = 38166.0 \text{ MJD}$$

$$\omega = (186^\circ 062 \pm 3) + (4^\circ 8917 \pm 3)t + .000134t^2 + .1295 \cos \omega$$

$$\Omega = (165^\circ 226 \pm 2) - (3^\circ 2865 \pm 2)t + (.10 \times 10^{-4})t^2 + .0090 \cos \omega$$

$$i = (33^\circ 3529 \pm 4) + .000185t - .0077 \sin \omega$$

$$e = (.18870 \pm 1) - (.62 \times 10^{-5})t + .000452 \sin \omega$$

$$M = (.484411 \pm 8) + (11.0884687 \pm 7)t - (.73 \pm 2) \times 10^{-6}t^2 \\ + (.54 \pm 2) \times 10^{-7}t^3 - .000376 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.15$.

The following elements are based on 151 observations and are valid for the period **June** 1 through July 1, 1963.

$$T_0 = 38196.0 \text{ MJD}$$

$$\omega = (332^\circ 855 \pm 3) + (4^\circ 8940 \pm 3)t + 0.000134t^2 + 0.1295 \cos \omega$$

$$\Omega = (66^\circ 618 \pm 2) - (3^\circ 2873 \pm 2)t + (0.10 \times 10^{-4})t^2 + 0.0090 \cos \omega$$

$$i = (33^\circ 3567 \pm 7) + 0.000185t - 0.0077 \sin \omega$$

$$e = (.18857 \pm 1) - .62 \times 10^{-5}t + .000452 \sin \omega$$

$$M = (.139165 \pm 9) + (11.0885196 \pm 8)t - (.19 \pm 2) \times 10^{-6}t^2 \\ - (.43 \pm 3) \times 10^{-7}t^3 - .000376 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.23$.

T (MJD)	ω	Ω	i	e	M	n	$n'/2$	q	N	D	σ
38122.0	330.84 2	309.876 9	33.365 4	.18856 3	.59157 3	11.088488 2	.38E-6 95	6.891868	16 6	.45	
38126.0	350.43 1	296.727 6	33.355 3	.18864 4	.94542 3	11.088504 5	.30E-5 19	6.891204	14 6	.54	
38130.0	1C.06 1	283.55 1	33.347 4	.18891 7	.29923 2	11.088504 4	-.37E-5 23	6.888941	14 6	.40	
38134.0	29.56 1	270.40 1	33.344 4	.18903 6	.65338 3	11.088499 5	-.64E-5 22	6.887895	23 6	.72	
38138.0	49.145 7	257.277 7	33.346 2	.18911 4	.00702 2	11.088481 3	-.32E-5 13	6.887234	25 6	.41	
38142.0	68.672 5	244.112 5	33.343 2	.18917 3	.36100 1	11.088467 1	.18E-6 63	6.886750	26 6	.42	
38146.0	88.181 4	230.967 4	33.344 1	.18922 3	.714984 9	11.088470 2	.4E-6 10	6.886285	24 6	.38	
38150.0	107.702 5	217.816 4	33.345 1	.18919 3	.06897 1	11.088473 1	.15E-5 9	6.886575	26 6	.42	
38154.0	127.245 8	204.662 6	33.349 2	.18914 3	.42295 2	11.088484 2	-.30E-5 9	6.886946	23 6	.49	
38158.0	146.783 8	191.509 6	33.349 1	.18896 3	.77698 2	11.088487 1	.14E-5 7	6.888458	24 6	.42	
38162.0	166.354 9	178.366 5	33.351 1	.18871 4	.13089 3	11.088482 3	-.14E-5 11	6.890596	45 6	.47	
38166.0	185.927 6	165.221 3	33.3546 9	.18863 4	.48480 3	11.088463 1	-.20E-5 9	6.891322	80 6	.46	
38170.0	205.518 5	152.073 2	33.3569 8	.18845 3	.83858 2	11.088457 1	.24E-5 7	6.892844	84 6	.43	
38174.0	225.117 5	138.927 2	33.3568 8	.18834 3	.19240 2	11.088462 1	-.42E-5 93	6.893756	71 6	.38	
38178.0	244.729 3	125.779 3	33.360 1	.18826 2	.546194 9	11.088472 2	.29E-5 9	6.894486	75 6	.44	
38182.0	264.351 5	112.635 4	33.359 2	.18820 3	.90002 1	11.088489 2	.27E-5 10	6.894919	49 6	.54	
38186.0	283.963 6	99.493 5	33.361 2	.18812 4	.25390 2	11.088486 2	-.10E-5 14	6.895605	27 6	.52	
38190.0	303.586 9	86.346 5	33.362 1	.18816 4	.60780 2	11.088496 2	-.24E-5 10	6.895251	24 6	.42	
38194.0	323.171 7	73.195 5	33.362 1	.18836 3	.96184 2	11.088496 1	-.21E-5 10	6.893584	30 6	.44	
38198.0	342.779 8	60.046 6	33.360 1	.18844 2	.31583 2	11.088499 1	.6E-7 82	6.892934	30 6	.45	
38202.0	2.35 1	46.916 8	33.356 2	.18857 3	.66989 3	11.088497 2	-.1E-6 11	6.891777	29 6	.54	
38206.0	21.94 1	33.356 7	33.356 3	.18870 3	.02396 2	11.088476 2	-.18E-5 10	6.890697	29 6	.49	
38210.0	41.48 1	20.608 6	33.356 4	.18884 4	.37802 3	11.088467 2	.1E-6 12	6.889549	31 6	.53	

Table 3

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1959 ETA

MJD	Z	φ	ψ	D. R.A.	\dot{P}
PERIGEE IN SUNLIGHT					
38122.	515.	-15.5	88.1	273.4	-0.618E-08
38126.	513.	-5.2	87.1	273.5	-0.488E-07
38130.	511.	5.5	86.1	273.1	0.602E-07
38134.	511.	15.7	84.4	273.2	0.104E-06
38138.	513.	24.6	81.0	275.1	0.521E-07
38142.	514.	30.8	76.0	279.1	-0.293E-08
38146.	514.	33.3	70.1	285.1	-0.651E-08
38150.	514.	31.6	64.4	291.2	-0.244E-07
38154.	513.	26.0	60.4	295.6	-0.488E-07
38158.	512.	17.5	59.3	297.6	-0.228E-07
38162.	513.	7.5	61.3	297.8	0.228E-07
38166.	513.	-3.3	65.7	297.1	0.325E-07
38170.	516.	-13.7	70.7	296.8	-0.390E-07
38174.	519.	-22.9	74.7	297.9	0.683E-08
38178.	521.	-29.8	76.3	301.2	-0.472E-07
38182.	523.	-33.2	75.0	306.7	-0.439E-07
38186.	523.	-32.3	71.0	312.8	0.163E-07
38190.	521.	-27.3	64.9	317.4	0.390E-07
38194.	518.	-19.2	57.9	319.6	-0.342E-07
38198.	515.	-9.4	51.2	319.8	-0.976E-09
38202.	513.	1.3	45.5	319.0	0.163E-08
38206.	513.	11.9	41.6	317.9	0.293E-07
38210.	514.	21.4	38.0	318.9	-0.163E-08

I. SAO smoothed elements

The following elements are based on 77 observations and are valid for the period April 1 through May 1, 1963.

$$T_0 = 38135.0 \text{ MJD}$$

$$\omega = (91^\circ 31 \pm 1) + (3^\circ 399 \pm 2)t + .940 \times 10^{-3}t^2 + 1^\circ 2206 \cos \omega$$

$$\Omega = (187^\circ 771 \pm 2) - (4^\circ 1914 \pm 3)t + .10 \times 10^{-6}t^2 + .0054 \cos \omega$$

$$i = (50^\circ 309 \pm 1) - .0014 \sin \omega$$

$$e = (.03621 \pm 2) - (.6 \pm 17) \times 10^{-6}t + .000790 \sin \omega$$

$$M = (.54571 \pm 4) + (14.234705 \pm 5)t - (.176 \pm 4) \times 10^{-5}t^2 \\ - (.76 \pm 42) \times 10^{-8}t^3 - .003059 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.73$.

The following elements are based on 64 observations and are valid for the period May 1 through June 1, 1963.

$$T_0 = 38165.0 \text{ MJD}$$

$$\omega = (193^\circ 28 \pm 4) + (3^\circ 413 \pm 4)t + .940 \times 10^{-3}t^2 + 1^\circ 2206 \cos \omega$$

$$\Omega = (62^\circ 036 \pm 2) - (4^\circ 1909 \pm 1)t + .10 \times 10^{-6}t^2 + .0054 \cos \omega$$

$$i = (50^\circ 306 \pm 2) - .0014 \sin \omega$$

$$e = (.03610 \pm 1) + (.21 \pm 11) \times 10^{-5}t + .000790 \sin \omega$$

$$M = (.5877 \pm 1) + (14.234713 \pm 9)t - (.199 \pm 5) \times 10^{-5}t^2 \\ + (.62 \pm 4) \times 10^{-7}t^3 - .003059 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.58$.

The following elements are based on 77 observations and are valid for the period June 1 through July 1, 1963.

$$T_0 = 38196.0 \text{ MJD}$$

$$\omega = (299^\circ 09 \pm 3) + (3^\circ 409 \pm 3)t + 9^\circ 940 \times 10^{-3}t^2 + 1^\circ 2206 \cos \omega$$

$$\Omega = (292^\circ 100 \pm 4) - (4^\circ 1920 \pm 5)t + 1^\circ 10 \times 10^{-6}t^2 + 9^\circ 0054 \cos \omega$$

$$i = (50^\circ 303 \pm 4) - 9^\circ 0014 \sin \omega$$

$$e = (.03604 \pm 6) + (.29 \pm 49) \times 10^{-5}t + .000790 \sin \omega$$

$$M = (.86514 \pm 9) + (14.234828 \pm 9)t - (.50 \pm 3) \times 10^{-6}t^2 \\ - (.60 \pm 5) \times 10^{-7}t^3 - .003059 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1^\circ 75$.

The following elements are based on 118 observations and are valid for the period July 1 through August 1, 1963.

$$T_0 = 38226.0 \text{ MJD}$$

$$\omega = (41^\circ 38 \pm 2) + (3^\circ 408 \pm 2)t + 9^\circ 000225t^2 + 1^\circ 2206 \cos \omega$$

$$\Omega = (166^\circ 365 \pm 2) - (4^\circ 1916 \pm 2)t - 9^\circ 13 \times 10^{-4}t^2 + 9^\circ 0054 \cos \omega$$

$$i = (50^\circ 305 \pm 1) - 9^\circ 0014 \sin \omega$$

$$e = (.03613 \pm 2) + (.68 \pm 18) \times 10^{-5}t + .000790 \sin \omega$$

$$M = (.91122 \pm 4) + (14.234958 \pm 6)t + (.88 \pm 4) \times 10^{-6}t^2 \\ - (.96 \pm 5) \times 10^{-7}t^3 - .003059 \cos \omega$$

Standard error of one observation: $\sigma = \pm 2^\circ 23$.

The following elements are based on 84 observations and are valid for the period August 1 through September 1, 1963.

$$T_0 = 38256.0 \text{ MJD}$$

$$\omega = (143^\circ 41 \pm 2) + (3^\circ 411 \pm 3)t + 9^\circ 000225t^2 + 1^\circ 2206 \cos \omega$$

$$\Omega = (40^\circ 620 \pm 1) - (4^\circ 1920 \pm 3)t - 9^\circ 13 \times 10^{-4}t^2 + 9^\circ 0054 \cos \omega$$

$$i = (50^\circ 309 \pm 2) - 9^\circ 0014 \sin \omega$$

$$e = (.03618 \pm 2) - (.4 \pm 19) \times 10^{-6}t + .000790 \sin \omega$$

$$M = (.96205 \pm 6) + (14.235052 \pm 8)t + (.54 \pm 5) \times 10^{-6}t^2 \\ + (.109 \pm 4) \times 10^{-6}t^3 - .003059 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1^\circ 83$.

II. SAO mean elements -- Satellite 1959 Iota 1

3 April - 30 June 1963

T (MJD)	ω	Ω	i	e	M	n	$n'/2$	q	N	D	σ
38122.0	47.81 5	242.24 1	50.295 7	.03674 3	.4931 1	14.234809 4	-.14E-5 10	6.927240	22	8	.73
38126.0	61.10 7	225.49 2	50.308 1	.03681 5	.4325 2	14.234823 4	-.30E-5 16	6.926731	15	8	1.14
38130.0	74.65 3	208.741 7	50.308 2	.03712 5	.37131 9	14.234825 2	.33E-5 10	6.924501	13	8	.48
38134.0	87.83 4	191.971 4	50.305 2	.03709 3	.3112 1	14.234827 3	-.22E-5 12	6.924758	23	8	.23
38138.0	101.1 2	175.203 6	50.305 4	.03706 9	.2510 6	14.234802 1	-.37E-5 6	6.924998	32	8	.47
38142.0	114.62 3	158.434 6	50.307 5	.03683 3	.18984 7	14.234776 2	-.37E-5 5	6.926652	26	8	.50
38146.0	127.95 4	141.666 3	50.306 2	.03677 4	.1294 1	14.234759 2	-.34E-6 59	6.927061	19	8	.72
38150.0	141.34 5	124.900 2	50.306 2	.03663 3	.0688 1	14.234757 2	-.75E-6 89	6.928044	20	8	.75
38154.0	154.74 9	108.134 2	50.307 3	.03646 3	.0083 2	14.234767 2	-.1E-7 80	6.929294	13	8	.58
38158.0	168.25 7	91.363 4	50.307 4	.03633 2	.9475 2	14.234743 2	-.51E-5 9	6.930255	10	8	.51
38162.0	181.97 9	74.600 8	50.307 4	.03610 4	.8861 3	14.234717 4	-.22E-5 14	6.931889	15	8	.70
38166.0	195.2 2	57.87 2	50.29 1	.0353 4	.8259 5	14.234694 2	-.33E-5 8	6.937608	26	8	.62
38170.0	209.1 2	41.08 1	50.305 7	.0354 5	.7642 5	14.234681 2	-.96E-6 82	6.937082	26	8	.63
38174.0	223.28 7	24.29 1	50.32 1	.03572 4	.7020 2	14.234668 6	-.11E-5 16	6.934673	13	8	.66
38178.0	237.04 5	7.546 7	50.313 6	.03547 2	.6407 1	14.234690 2	-.19E-5 9	6.936451	12	8	.56
38182.0	250.98 5	350.786 3	50.308 3	.03540 1	.5790 1	14.234664 2	-.16E-5 8	6.936923	23	8	.66
38186.0	264.87 4	334.016 2	50.313 2	.03532 1	.5175 1	14.234644 1	-.38E-5 5	6.937507	18	8	.45
38190.0	278.60 9	317.245 4	50.299 8	.03537 5	.4564 2	14.234630 2	-.17E-5 11	6.937188	14	8	.60
38194.0	292.67 8	300.490 8	50.307 9	.0356 4	.3944 3	14.234640 2	-.1E-5 1	6.935795	18	8	.65
38198.0	306.55 5	283.371 9	50.300 7	.0356 4	.3331 2	14.234645 2	-.49E-6 77	6.935810	27	8	.64
38202.0	320.63 4	266.920 9	50.281 6	.0349 2	.2716 1	14.234636 1	-.20E-5 5	6.940516	27	8	.45
38206.0	334.7 1	250.15 2	50.28 1	.0348 3	.2100 3	14.234614 2	-.25E-5 9	6.941597	22	8	.69
38210.0	348.0 2	233.423 5	50.304 6	.03587 5	.1499 4	14.234584 3	-.29E-5 11	6.933585	19	8	.90

T (MJD)	ω	Ω	i	e	M	n	$n'/2$	q	N	D	σ
38214.0	1.87 4	216.666 3	50.302 2	.03607 2	.0887 1	14.235046 2	.62E-5 8	6.931979	34	8	.91
38218.0	15.28 3	199.902 2	50.305 1	.03638 2	.02878 7	14.225099 1	.76E-5 6	6.930504	48	8	.79
38222.0	28.71 3	183.135 3	50.306 2	.03638 3	.96907 7	14.235132 2	.24E-5 8	6.929743	38	8	.87
38226.0	42.18 3	166.374 3	50.305 2	.03650 4	.90931 8	14.235134 2	-.17E-6 8	6.928850	32	8	.85
38230.0	55.61 3	149.604 4	50.306 2	.03676 4	.84960 9	14.235124 2	-.17E-5 8	6.927012	35	8	.76
38234.0	68.96 3	132.829 5	50.308 2	.03693 3	.79012 8	14.235116 2	-.48E-6 75	6.925805	29	8	.58
38238.0	82.25 4	116.074 1	50.302 6	.03697 4	.7308 1	14.235131 2	.56E-5 9	6.925534	19	8	.71
38242.0	95.66 3	99.314 5	50.295 4	.03686 3	.67136 7	14.235162 2	.18E-5 9	6.926293	16	8	.46
38246.0	108.94 4	82.535 4	50.307 5	.03684 6	.6124 1	14.235203 2	.57E-5 10	6.926434	16	8	.61
38250.0	122.33 4	65.770 3	50.308 4	.03680 4	.5533 1	14.235216 2	.8E-6 11	6.926712	27	8	1.01
38254.0	135.72 2	48.998 2	50.310 2	.03669 2	.49416 7	14.235214 2	-.28E-5 6	6.927507	34	8	.58
38258.0	149.08 4	32.226 2	50.308 3	.03655 2	.4351 1	14.235205 1	-.49E-6 67	6.928475	22	8	.58
38262.0	162.72 3	15.456 3	50.297 3	.03645 1	.37533 9	14.235219 1	.51E-5 5	6.929217	20	8	.38
38266.0	176.18 4	358.691 6	50.305 5	.03624 2	.31622 1	14.235258 1	.54E-5 6	6.930690	25	8	.48
38270.0	190.1 1	341.91 1	50.305 7	.03603 5	.2561 3	14.235286 2	.23E-5 1	6.9322191	18	8	.78
38274.0	203.63 8	325.173 5	50.309 3	.03584 3	.1969 2	14.235296 7	-.33E-5 19	6.933587	16	8	.63

Table 4

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1959 IOTA 1

MJD	Z	φ	ψ	D. R.A.	\dot{P}
PERIGEE IN SUNLIGHT					
38122.	556.	34.8	90.5	265.9	0.138E-07
38126.	558.	42.4	93.3	259.5	-0.296E-07
38130.	558.	47.9	92.9	256.6	-0.326E-07
38134.	559.	50.3	91.5	256.1	0.217E-07
38138.	559.	49.0	90.8	256.0	0.365E-07
38142.	559.	44.4	92.5	254.1	0.365E-07
38146.	557.	37.4	98.0	248.6	0.336E-08
38150.	555.	28.7	107.3	240.3	-0.740E-08
PERIGEE IN EARTH SHADOW					
38154.	553.	19.2	119.5	230.0	0.987E-10
38158.	552.	9.0	133.8	218.6	0.503E-07
38162.	554.	-1.5	148.9	206.7	0.217E-07
38166.	560.	-11.6	164.0	194.7	0.326E-07
38170.	562.	-22.0	176.1	183.6	0.948E-08
38174.	562.	-31.8	167.8	174.3	0.109E-07
38178.	567.	-40.2	158.2	167.0	0.188E-07
38182.	570.	-46.7	151.9	163.2	0.158E-07
38186.	572.	-50.0	149.4	162.7	0.375E-07
38190.	571.	-49.5	150.3	163.2	-0.168E-07
38194.	568.	-45.2	153.6	162.1	-0.987E-08
38198.	566.	-38.2	155.4	156.9	0.484E-08
38202.	567.	-29.2	152.0	149.4	0.197E-07

Table 4 (cont.)

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1959 IOTA 1

MJD	Z	φ	ψ	D. R.A.	\dot{P}
38206.	566.	-19.2	142.0	139.3	0.247E-07
38210.	556.	-9.2	128.0	127.5	0.286E-07
PERIGEE IN SUNLIGHT					
38214.	554.	1.4	112.8	115.5	-0.612E-07
38218.	553.	11.7	97.5	103.4	-0.750E-07
38222.	554.	21.7	83.6	91.9	-0.237E-07
38226.	556.	31.1	72.4	81.8	0.168E-08
38230.	557.	39.4	64.9	74.0	0.168E-07
38234.	558.	45.9	61.4	69.2	0.474E-08
38238.	560.	49.7	61.0	67.5	-0.553E-07
38242.	560.	50.0	61.8	67.7	-0.178E-07
38246.	559.	46.7	61.5	66.4	-0.563E-07
38250.	557.	40.6	58.7	62.3	-0.790E-08
38254.	555.	32.5	52.8	55.1	0.276E-07
38258.	553.	23.3	44.1	45.6	0.484E-08
38262.	552.	13.2	33.9	34.8	-0.503E-07
38266.	552.	2.9	24.3	23.1	-0.533E-07
38270.	554.	-7.8	20.9	11.6	-0.227E-07
38274.	557.	-18.0	26.2	0.4	0.326E-07

T (MTD)	ω	Ω	i	e	M	n	n'/2	q	N	D	σ
38120.0	9. 1	52.58 1	47.203 7	.00417 9	.707 3	12.46568 2	-.9E-4 4	7.823838	34	2	1.23
38121.0	16. 2	49.29 2	47.21 1	.0042 2	.165 7	12.46592 3	-.20E-3 6	7.823579	31	2	1.89
38122.0	28.3 9	46.05 1	47.195 7	.0051 1	.607 3	12.46608 3	-.6E-4 5	7.816298	29	2	1.41
38123.0	33. 1	42.78 2	47.20 1	.0054 2	.071 3	12.46630 3	.14E-3 5	7.81	26	2	1.98
38124.0	42.3 6	39.51 1	47.196 6	.0062 1	.522 2	12.46657 2	-.2E-4 5	7.307612	25	2	1.30
38125.0	46.7 6	36.26 1	47.202 6	.0066 1	.987 2	12.46677 3	.14E-3 5	7.804178	27	2	1.36
38126.0	52.0 3	32.979 8	47.197 3	.00749 7	.4502 9	12.46693 1	.9E-4 2	7.797214	20	2	.65
38127.0	59.7 3	29.724 9	47.188 4	.00827 9	.9063 9	12.46723 1	.15E-3 3	7.790958	15	2	.54
38128.0	63.6 7	26.45 3	47.191 9	.0087 2	.374 2	12.46750 3	.41E-3 8	7.787724	12	2	1.37
38129.0	68.6 2	23.175 7	47.186 3	.00939 6	.8381 5	12.46782 1	.15E-3 2	7.781901	15	2	.51
38130.0	73.5 2	19.90 1	47.188 4	.01012 9	.3031 5	12.46809 1	.5E-4 3	7.776095	24	2	.91
38131.0	77.5 2	16.65 1	47.187 4	.01072 8	.7707 5	12.46839 2	-.22E-3 3	7.771200	29	2	.94
38132.0	81.6 1	13.385 7	47.188 2	.01148 4	.2385 3	12.46884 1	.24E-3 2	7.765092	24	2	.40
38133.0	86.4 1	10.102 8	47.184 3	.01224 5	.7049 3	12.469152 9	.22E-3 2	7.758956	23	2	.53
38134.0	90.20 8	6.841 6	47.181 3	.01310 5	.1742 2	12.469573 8	.20E-3 1	7.752013	22	2	.49
38135.0	94.35 7	3.561 6	47.192 3	.01375 5	.6431 2	12.47002 1	.18E-3 2	7.746758	19	2	.42
38136.0	98.11 7	0.298 6	47.194 4	.01455 6	.1133 2	12.470410 8	.19E-3 2	7.740284	20	2	.53
38137.0	101.90 4	357.034 5	47.186 3	.01512 4	.5840 2	12.470851 6	.21E-3 1	7.735608	24	2	.46
38138.0	105.76 4	353.766 4	47.188 3	.01592 3	.0548 1	12.471310 9	.24E-3 1	7.729175	22	2	.34
38139.0	109.47 6	350.492 5	47.194 4	.01669 6	.5266 2	12.471762 6	.23E-3 1	7.722907	17	2	.45
38140.0	113.23 7	347.224 6	47.191 5	.01733 6	.9986 2	12.47226 1	.27E-3 2	7.717668	25	2	.66
38141.0	116.91 7	343.958 6	47.196 6	.01809 6	.4713 2	12.47273 1	.26E-3 2	7.711534	26	2	.71
38142.0	120.51 5	340.681 4	47.204 5	.01878 6	.9447 2	12.47323 2	.25E-3 2	7.705887	18	2	.42
38143.0	124.29 8	337.412 6	47.211 5	.01953 7	.4181 2	12.47370 2	.22E-3 3	7.699816	16	2	.49
38144.0	127.9 1	334.148 9	47.212 7	.0202 1	.8925 3	12.47418 2	.24E-3 3	7.694447	13	2	.73
38145.0	131.41 7	330.874 4	47.207 4	.02057 6	.3676 2	12.47472 2	.26E-3 2	7.691251	13	2	.44
38146.0	135.17 5	327.616 4	47.212 3	.02132 5	.8425 2	12.475210 7	.24E-3 1	7.685159	18	2	.44
38147.0	138.92 5	324.347 4	47.215 4	.02204 5	.3179 2	12.475747 8	.29E-3 1	7.679252	16	2	.43
38148.0	142.57 6	321.085 4	47.211 4	.02258 6	.7942 2	12.47629 1	.24E-3 2	7.674795	15	2	.60
38149.0	146.29 4	317.815 2	47.213 3	.02326 4	.2707 1	12.476797 8	.27E-3 1	7.669240	21	2	.48

T (MJD)	ω	Ω	i	e	M	n	n'/2	q	N	D	σ
38150.0	150.02 7	314.541 3	47.214 3	.02383 6	.7478 2	12.477363 9	.27E-3 2	7.664590	16 2	.54	
38151.0	153.7 1	311.274 4	47.217 4	.02435 8	.2257 3	12.47795 2	.29E-3 4	7.660271	10 2	.61	
38152.0	157.38 9	308.005 3	47.216 3	.02491 5	.7039 3	12.47849 1	.23E-3 2	7.655633	14 2	.48	
38153.0	161.39 5	304.732 3	47.219 3	.02564 4	.1818 2	12.479065 8	.27E-3 2	7.649681	21 2	.58	
38154.0	165.18 6	301.464 3	47.218 3	.02621 5	.6609 2	12.47961 1	.25E-3 2	7.644953	20 2	.61	
38155.0	168.95 5	298.190 3	47.213 3	.02676 5	.1406 1	12.480174 6	.28E-3 1	7.640401	16 2	.45	
38156.0	172.66 6	294.919 3	47.215 3	.02726 5	.6210 2	12.48077 1	.30E-3 2	7.636258	13 2	.45	
38157.0	176.65 9	291.648 5	47.214 5	.02794 6	.1012 3	12.48132 1	.26E-3 2	7.630691	13 2	.68	
38158.0	180.4 1	288.372 7	47.215 6	.02854 7	.5825 3	12.48194 2	.33E-3 3	7.625720	11 2	.72	
38159.0	184.3 1	285.087 9	47.222 2	.02891 9	.0642 3	12.48256 2	.24E-3 4	7.622527	8 2	.61	
38160.0	188.0 2	281.82 2	47.222 1	.0296 1	.5470 4	12.48311 3	.33E-3 5	7.616909	10 2	.91	
38161.0	192.0 1	278.55 1	47.211 9	.03031 9	.0295 3	12.48380 2	.23E-3 3	7.611071	15 2	.75	
38162.0	195.72 7	275.274 9	47.206 7	.03088 6	.5133 2	12.48438 2	.27E-3 3	7.606384	16 2	.61	
38163.0	199.55 8	271.99 1	47.208 8	.03145 8	.9975 2	12.48500 1	.29E-3 2	7.601621	12 2	.64	
38164.0	203.43 6	268.704 6	47.201 5	.03218 5	.4822 2	12.48562 1	.33E-3 2	7.595684	13 2	.41	
38165.0	207.18 4	265.424 5	47.201 4	.03276 4	.9679 1	12.486270 6	.32E-3 1	7.590825	22 2	.44	
38166.0	210.90 4	262.144 4	47.206 3	.03334 3	.4542 1	12.486896 6	.31E-3 1	7.586063	27 2	.46	
38167.0	214.64 5	258.862 6	47.210 4	.03396 5	.9411 1	12.487509 8	.30E-3 2	7.580922	22 2	.53	
38168.0	218.39 6	255.572 7	47.205 6	.03467 7	.4286 2	12.488132 9	.34E-3 2	7.575089	25 2	.67	
38169.0	221.99 4	252.287 5	47.205 4	.03530 4	.9172 1	12.488775 6	.33E-3 1	7.569866	23 2	.44	
38170.0	225.68 5	248.999 9	47.205 6	.03598 5	.4061 1	12.489433 9	.33E-3 2	7.564266	20 2	.52	
38171.0	229.31 6	245.714 9	47.211 6	.03665 7	.8958 2	12.49006 1	.33E-3 2	7.558803	17 2	.61	
38172.0	232.89 4	242.430 7	47.215 4	.03725 6	.3863 1	12.49071 1	.29E-3 2	7.553797	20 2	.52	
38173.0	236.50 4	239.128 7	47.212 4	.03788 5	.8773 1	12.49133 1	.31E-3 2	7.548634	27 2	.55	
38174.0	239.89 6	235.84 1	47.216 6	.03848 7	.3696 2	12.49194 2	.33E-3 3	7.543679	32 2	.80	
38175.0	243.48 4	232.560 7	47.221 4	.03922 5	.8619 1	12.492602 8	.35E-3 2	7.537588	29 2	.60	
38176.0	247.03 4	229.264 8	47.223 3	.03990 6	.3550 1	12.49325 1	.32E-3 2	7.532010	27 2	.61	
38177.0	250.51 3	225.968 6	47.227 3	.04056 5	.8489 1	12.49390 1	.36E-3 2	7.526538	32 2	.69	
38178.0	253.89 2	222.674 3	47.229 2	.04125 3	.34370 5	12.494576 8	.34E-3 1	7.520837	44 2	.54	
38179.0	257.38 2	219.379 3	47.232 1	.04187 4	.83886 5	12.495246 6	.36E-3 1	7.515722	52 2	.64	
38180.0	260.88 2	216.083 2	47.235 1	.04246 3	.33466 5	12.495862 6	.262E-3 9	7.510816	42 2	.49	

T (MJD)	w	Ω	i	e	M	n	n'/2	q	N	D	σ
38181.0	264.19	3	212.788	4	.47-240 2	.04312 6	.83158 9	12.496514 8	.34E-3 1	7.505452	.39 2
38182.0	267.54	3	209.489	4	.47-249 2	.04367 6	.32899 9	12.497131 9	.31E-3 2	7.500827	.36 2
38183.0	270.86	3	206.202	4	.47-256 3	.04437 6	.82703 9	12.49767 1	.35E-3 2	7.495180	.34 2
38184.0	274.32	3	202.906	4	.47-260 3	.04490 6	.32532 8	12.498198 9	.21E-3 1	7.490804	.25 2
38185.0	277.65	4	199.609	4	.47-262 4	.04541 7	.8243 1	12.49862 1	.22E-3 2	7.486598	.28 2
38186.0	281.02	2	196.309	3	.47-261 2	.04591 4	.32365 7	12.498905 7	.10E-3 1	7.482604	.28 2
38187.0	284.37	3	193.008	3	.47-260 3	.04647 5	.82316 7	12.499152 7	.16E-3 1	7.478121	.25 2
38188.0	287.78	3	189.711	3	.47-259 3	.04685 5	.32281 9	12.499433 9	.15E-3 1	7.475042	.27 2
38189.0	291.18	3	186.411	3	.47-263 3	.04747 4	.82270 9	12.499670 5	.11E-3 1	7.470080	.30 2
38190.0	294.55	2	183.111	3	.47-265 1	.04780 3	.32298 6	12.499926 5	.111E-3 8	7.467336	.41 2
38191.0	298.04	2	179.811	3	.47-265 1	.04811 3	.82313 6	12.500142 6	.100E-3 8	7.464848	.41 2
38192.0	301.41	3	176.511	3	.47-266 1	.04842 3	.3238 1	12.500400 4	.136E-3 8	7.462334	.41 2
38193.0	304.93	3	173.215	3	.47-267 1	.04873 3	.8244 1	12.500648 5	.129E-3 8	7.459774	.42 2
38194.0	308.42	3	169.909	3	.47-268 1	.04895 2	.3252 1	12.500913 5	.135E-3 9	7.457977	.31 2
38195.0	311.87	3	166.613	3	.47-271 2	.04913 2	.82645 9	12.501179 6	.14E-3 1	7.456427	.36 2
38196.0	315.52	3	163.315	3	.47-272 2	.04930 2	.32735 9	12.501435 6	.17E-3 1	7.455009	.47 2
38197.0	318.96	2	160.009	2	.47-274 1	.04932 1	.82916 7	12.501700 4	.109E-3 7	7.454768	.48 2
38198.0	322.67	3	156.711	2	.47-274 2	.04946 1	.33037 9	12.502007 5	.155E-3 8	7.453528	.49 2
38199.0	326.34	3	153.408	2	.47-277 2	.04953 1	.83199 8	12.502315 5	.122E-3 9	7.452839	.51 2
38200.0	330.16	3	150.109	2	.47-278 2	.04959 1	.33345 9	12.502570 6	.11E-3 1	7.452250	.46 2
38201.0	333.83	5	146.809	4	.47-284 3	.04963 3	.8356 2	12.502826 8	.14E-3 2	7.451869	.48 2
38202.0	337.99	4	143.510	3	.47-282 2	.04960 2	.3365 1	12.50316 5	.4E-4 5	7.451935	.34 2
38203.0	341.40	4	140.212	3	.47-284 2	.04966 2	.8399 1	12.503293 6	.9E-4 1	7.451441	.53 2
38204.0	345.57	5	136.917	3	.47-284 2	.04968 2	.3413 1	12.503517 6	.5E-4 1	7.451165	.41 2
38205.0	349.31	5	133.617	3	.47-283 2	.04974 2	.8440 2	12.503673 7	.5E-4 1	7.450670	.42 2
38206.0	353.27	5	130.314	3	.47-278 2	.04977 2	.3464 1	12.503878 7	.10E-3 1	7.450347	.50 2
38207.0	357.03	5	127.013	4	.47-279 2	.04982 2	.8495 2	12.504074 8	.4E-4 1	7.449877	.46 2
38208.0	1.06	5	123.716	3	.47-275 2	.04981 2	.3519 2	12.504274 8	.5E-4 1	7.449899	.41 2
38209.0	4.90	5	120.416	3	.47-267 2	.04988 2	.8551 1	12.504437 6	.9E-4 1	7.449290	.39 2
38210.0	8.87	8	117.109	6	.47-266 3	.04995 3	.3582 2	12.50465 1	.9E-4 2	7.448638	.37 2

Table 5

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1960 IOTA 1

MJD	Z	φ	ψ	D. R.A.	\dot{P}
PERIGEE IN SUNLIGHT					
38126.	1446.	6.6	48.8	49.0	0.116E-05
38127.	1446.	11.7	49.6	49.7	0.257E-05
38122.	1441.	20.4	55.2	54.6	0.772E-06
38123.	1439.	23.6	55.1	54.1	-0.180E-05
38124.	1434.	29.6	59.4	57.9	0.257E-06
38125.	1432.	32.3	59.7	57.8	-0.180E-05
38126.	1426.	35.3	61.0	58.8	-0.116E-05
38127.	1421.	39.3	64.8	62.9	-0.193E-05
38128.	1419.	41.1	65.2	63.3	-0.528E-05
38129.	1414.	43.1	66.9	65.3	-0.193E-05
38130.	1408.	44.7	68.5	67.5	-0.643E-06
38131.	1404.	45.7	69.3	68.8	-0.283E-05
38132.	1398.	46.5	70.3	70.4	-0.309E-05
38133.	1392.	47.1	72.0	73.2	-0.283E-05
38134.	1385.	47.2	72.7	74.6	-0.257E-05
38135.	1380.	47.0	73.7	76.5	-0.232E-05
38136.	1373.	46.6	74.3	77.8	-0.244E-05
38137.	1368.	45.9	74.8	79.0	-0.270E-05
38138.	1361.	44.9	75.4	80.1	-0.309E-05
38139.	1355.	43.8	75.7	80.8	-0.296E-05
38140.	1349.	42.4	75.9	81.4	-0.347E-05
38141.	1342.	40.9	76.0	81.7	-0.334E-05

Table 5 (cont.)

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1960 IOTA 1

MJD	Z	ϕ	ψ	D.R.A.	\dot{P}
38142.	1336.	39.2	75.9	81.7	-0.321E-05
38143.	1329.	37.3	75.8	81.6	-0.283E-05
38144.	1323.	35.4	75.5	81.2	-0.308E-05
38145.	1319.	33.4	74.9	80.5	-0.334E-05
38146.	1313.	31.2	74.5	79.9	-0.308E-05
38147.	1306.	28.8	73.9	79.0	-0.373E-05
38148.	1301.	26.5	73.2	78.0	-0.308E-05
38149.	1294.	24.0	72.4	76.9	-0.347E-05
38150.	1289.	21.5	71.5	75.6	-0.347E-05
38151.	1284.	19.0	70.6	74.2	-0.373E-05
38152.	1279.	16.4	69.6	72.8	-0.295E-05
38153.	1272.	13.5	68.8	71.5	-0.347E-05
38154.	1267.	10.8	67.9	69.9	-0.321E-05
38155.	1262.	8.1	67.0	68.3	-0.360E-05
38156.	1258.	5.4	66.1	66.6	-0.385E-05
38157.	1252.	2.5	65.5	65.1	-0.334E-05
38158.	1247.	-0.3	64.8	63.4	-0.424E-05
38159.	1244.	-3.2	64.3	61.8	-0.308E-05
38160.	1239.	-5.9	63.8	60.1	-0.424E-05
38161.	1233.	-8.8	63.7	58.6	-0.295E-05
38162.	1229.	-11.5	63.5	57.0	-0.346E-05
38163.	1225.	-14.2	63.6	55.5	-0.372E-05
38164.	1219.	-17.0	64.0	54.0	-0.423E-05
38165.	1215.	-19.6	64.3	52.6	-0.411E-05
38166.	1211.	-22.1	64.9	51.2	-0.398E-05
38167.	1206.	-24.7	65.6	50.0	-0.385E-05

Table 5 (cont.)

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1960 IOTA 1

MJD	Z	φ	ψ	D. R.A.	P
38168.	1201.	-27.1	66.4	48.8	-0.436E-05
38169.	1197.	-29.4	67.3	47.7	-0.423E-05
38170.	1192.	-31.7	68.4	46.8	-0.423E-05
38171.	1187.	-33.8	69.5	46.0	-0.423E-05
38172.	1183.	-35.8	70.7	45.3	-0.372E-05
38173.	1178.	-37.7	71.9	44.8	-0.397E-05
38174.	1174.	-39.4	73.0	44.3	-0.423E-05
38175.	1168.	-41.1	74.3	44.2	-0.449E-05
38176.	1163.	-42.5	75.6	44.2	-0.410E-05
38177.	1158.	-43.8	76.7	44.4	-0.461E-05
38178.	1153.	-44.8	77.8	44.5	-0.436E-05
38179.	1148.	-45.8	78.8	45.0	-0.461E-05
38180.	1144.	-46.5	79.7	45.6	-0.336E-05
38181.	1139.	-46.9	80.4	46.1	-0.435E-05
38182.	1134.	-47.2	81.0	46.7	-0.397E-05
38183.	1128.	-47.2	81.4	47.3	-0.448E-05
38184.	1124.	-47.1	81.8	48.0	-0.269E-05
38185.	1120.	-46.7	81.9	48.5	-0.282E-05
38186.	1115.	-46.1	81.8	49.0	-0.128E-05
38187.	1111.	-45.4	81.5	49.4	-0.205E-05
38188.	1107.	-44.4	81.0	49.7	-0.192E-05
38189.	1102.	-43.2	80.3	49.8	-0.141E-05
38190.	1099.	-41.9	79.4	49.6	-0.142E-05
38191.	1095.	-40.4	78.4	49.5	-0.128E-05
38192.	1092.	-38.8	77.0	49.0	-0.174E-05

Table 5 (cont.)

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1960 IOTA 1

MJD	Z	φ	ψ	D. R.A.	\dot{P}
38193.	1089.	-37.0	75.6	48.5	-0.165E-05
38194.	1087.	-35.1	73.9	47.8	-0.173E-05
38195.	1084.	-33.2	72.0	46.9	-0.179E-05
38196.	1082.	-31.0	69.9	46.0	-0.218E-05
38197.	1081.	-28.8	67.7	44.8	-0.139E-05
38198.	1079.	-26.5	65.3	43.6	-0.198E-05
38199.	1078.	-24.0	62.8	42.3	-0.156E-05
38200.	1077.	-21.4	60.1	41.1	-0.141E-05
38201.	1076.	-18.9	57.3	39.5	-0.179E-05
38202.	1075.	-16.0	54.4	38.3	-0.512E-06
38203.	1074.	-13.6	51.4	36.4	-0.115E-05
38204.	1073.	-10.5	48.3	35.1	-0.640E-06
38205.	1073.	-7.8	45.1	33.3	-0.640E-06
38206.	1072.	-4.9	41.9	31.7	-0.128E-05
38207.	1072.	-2.2	38.7	29.9	-0.512E-06
38208.	1072.	0.8	35.5	28.3	-0.640E-06
38209.	1071.	3.6	32.4	26.6	-0.115E-05
38210.	1071.	6.5	29.3	25.0	-0.115E-05

I. SAO smoothed elements

The following elements are based on 100 observations and are valid for the period April 1 through May 1, 1963.

$$T_0 = 38135.0 \text{ MJD}$$

$$\omega = (41^\circ 955 \pm 7) + (2^\circ 8167 \pm 7)t + .55 \times 10^{-4}t^2 + .3431 \cos \omega$$

$$\Omega = (234^\circ 857 \pm 2) - (3^\circ 3899 \pm 2)t + .14 \times 10^{-5}t^2 + .0143 \cos \omega$$

$$i = (49^\circ 940 \pm 1) + .117 \times 10^{-4}t + .0043 \sin \omega$$

$$e = (.11900 \pm 2) - .46 \times 10^{-5}t + .0007285 \sin \omega$$

$$M = (.89567 \pm 2) + (12.811465 \pm 2)t + (.722 \pm 4) \times 10^{-5}t^2 \\ - (.26 \pm 4) \times 10^{-7}t^3 - .00087990 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1^\circ 58.$

The following elements are based on 101 observations and are valid for the period May 1 through June 1, 1963.

$$T_0 = 38165.0 \text{ MJD}$$

$$\omega = (126^\circ 396 \pm 6) + (2^\circ 8130 \pm 9)t + .55 \times 10^{-4}t^2 + .3431 \cos \omega$$

$$\Omega = (133^\circ 163 \pm 1) - (3^\circ 3902 \pm 2)t + .14 \times 10^{-5}t^2 + .0143 \cos \omega$$

$$i = (49^\circ 9463 \pm 9) + .117 \times 10^{-4}t - .0043 \sin \omega$$

$$e = (.119031 \pm 8) - .46 \times 10^{-5}t + .0007285 \sin \omega$$

$$M = (.24607 \pm 1) + (12.811920 \pm 2)t + (.732 \pm 4) \times 10^{-5}t^2 \\ - (.100 \pm 4) \times 10^{-6}t^3 - .00087990 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1^\circ 45.$

The following elements are based on 82 observations and are valid for the period June 1 through July 1, 1963.

$$T_0 = 38196.0 \text{ MJD}$$

$$\omega = (213^{\circ}677 \pm 9) + (2^{\circ}8188 \pm 9)t + .55 \times 10^{-4}t^2 + .3431 \cos \omega$$

$$\Omega = (1^{\circ}079 \pm 2) - (3^{\circ}3901 \pm 3)t + .14 \times 10^{-5}t^2 + .0143 \cos \omega$$

$$i = (49^{\circ}951 \pm 2) + .117 \times 10^{-4}t - .0043 \sin \omega$$

$$e = (.118907 \pm 9) - .46 \times 10^{-5}t + .0007285 \sin \omega$$

$$M = (.42103 \pm 2) + (12.812293 \pm 2)t + (.689 \pm 3) \times 10^{-5}t^2 \\ - (.43 \pm 4) \times 10^{-7}t^3 - .00087990 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.50$.

II. SAO mean elements -- Satellite 1960 Xi 1

3 April - 30 June 1963

Ω	ω	i	e	M	n	$n^1/2$	q	N	D	σ
(MJD)										
38122.0	5.668	6	278.944	2	49.945	2	.11909	2	.34709	1
38126.0	16.92	1	265.38	1	49.940	4	.11916	9	.59230	5
38130.0	28.20	2	251.812	6	49.940	4	.11940	6	.83773	4
38134.0	39.393	6	238.245	3	49.934	2	.11948	1	.08354	1
38138.0	5C.60	1	224.652	5	49.941	3	.11955	3	.32964	3
38142.0	61.81	1	211.137	6	49.943	2	.11966	3	.57598	3
38146.0	73.CC	1	197.572	7	49.943	4	.11965	5	.82254	3
38150.0	84.24	6	184.01	1	49.945	8	.1196	3	.0691	3
38154.0	95.45	6	170.456	8	49.945	8	.1196	2	.3160	2
38158.0	106.56	2	156.891	2	49.943	2	.11968	2	.56341	4
38162.0	117.75	1	143.331	2	49.944	1	.11963	2	.81090	3
38166.0	128.97	1	129.769	2	49.942	2	.11957	1	.05859	3
38170.0	140.23	1	116.206	5	49.948	5	.11943	2	.30645	4
38174.0	151.40	2	102.643	5	49.949	5	.11934	4	.55468	4
38178.0	162.62	2	89.067	9	49.946	3	.11921	2	.80296	6
38182.0	173.87	2	75.510	9	49.950	3	.11905	2	.05129	6
38186.0	185.17	1	61.971	6	49.947	4	.11885	1	.29968	4
38190.0	196.47	2	48.411	8	49.947	8	.11873	2	.54828	5
38194.0	207.73	2	34.852	5	49.948	6	.11858	2	.79724	4
38198.0	219.04	1	21.293	5	49.951	4	.11844	3	.04631	3
38202.0	230.37	1	7.730	2	49.955	2	.11838	1	.29559	3
38206.0	241.690	9	354.170	1	49.954	2	.11825	1	.54507	2
38210.0	253.01	1	340.612	3	49.956	4	.11821	3	.79471	3

Table 6

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1960 XI 1

MJD	Z	φ	ψ	D. R.A.	\dot{P}
PERIGEE IN SUNLIGHT					
38122.	418.	4.3	88.6	271.1	-0.829E-07
38126.	418.	12.9	97.0	261.3	-0.463E-07
38130.	418.	21.2	103.6	252.0	-0.536E-07
38134.	420.	29.1	107.7	243.6	-0.171E-06
38138.	422.	36.3	109.0	236.5	-0.116E-06
38142.	423.	42.4	107.9	231.4	-0.682E-07
38146.	425.	47.1	105.6	228.4	-0.134E-07
38150.	426.	49.6	103.2	227.6	-0.877E-07
38154.	426.	49.6	102.1	227.5	-0.113E-06
38158.	425.	47.2	103.4	226.5	-0.158E-06
38162.	423.	42.6	107.2	223.5	-0.113E-06
PERIGEE IN EARTH SHADOW					
38166.	421.	36.5	113.7	218.2	-0.950E-07
38170.	420.	29.3	122.3	211.0	-0.609E-08
38174.	418.	21.5	132.5	202.3	-0.268E-07
38178.	418.	13.2	143.2	192.6	-0.694E-07
38182.	418.	4.7	153.1	182.4	-0.682E-07
38186.	419.	-4.0	159.9	172.0	-0.573E-07
38190.	421.	-12.5	159.8	161.8	-0.987E-07
38194.	424.	-20.9	154.0	152.0	-0.107E-06
38198.	427.	-28.8	146.6	143.1	-0.116E-06
38202.	430.	-36.1	139.9	135.7	-0.366E-07
38206.	433.	-42.4	134.8	130.2	-0.780E-07
38210.	436.	-47.1	131.7	127.0	-0.341E-07

I. SAO smoothed elements

The following elements are based on 114 observations and are valid for the period April 1 through April 15, 1963.

$$T_0 = 38126.0 \text{ MJD}$$

$$\omega = (240^\circ 605 \pm 6) + (4^\circ 868 \pm 1)t + 99 \times 10^{-4}t^2 + 2484 \cos \omega$$

$$\Omega = (205^\circ 096 \pm 4) - (3^\circ 7325 \pm 7)t - 000248t^2 + 0066 \cos \omega$$

$$i = (38^\circ 888 \pm 1) + 000325t - 0045 \sin \omega$$

$$e = (.12391 \pm 3) + (.78 \pm 5) \times 10^{-4}t + .41 \times 10^{-6}t^2 + .0005258 \sin \omega$$

$$M = (.02223 \pm 2) + (12.288177 \pm 4)t + (.0002559 \pm 5)t^2 - (.22 \pm 1) \times 10^{-5}t^3 \\ + (.28 \pm 1) \times 10^{-6}t^4 - .0006974 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.83$.

The following elements are based on 182 observations and are valid for the period April 15 through May 1, 1963.

$$T_0 = 38141.0 \text{ MJD}$$

$$\omega = (313^\circ 638 \pm 3) + (4^\circ 8758 \pm 6)t + 99 \times 10^{-4}t^2 + 2484 \cos \omega$$

$$\Omega = (149^\circ 051 \pm 2) - (3^\circ 7398 \pm 3)t - 000248t^2 + 0066 \cos \omega$$

$$i = (38^\circ 8980 \pm 6) + 000325t - 0045 \sin \omega$$

$$e = (.12505 \pm 2) + (.55 \pm 3) \times 10^{-4}t + .41 \times 10^{-6}t^2 + .0005258 \sin \omega$$

$$M = (.404369 \pm 8) + (12.296080 \pm 2)t + (.0002319 \pm 2)t^2$$

$$- (.271 \pm 4) \times 10^{-5}t^3 + (.71 \pm 4) \times 10^{-7}t^4 - .0006974 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.38$.

The following elements are based on 135 observations and are valid for the period May 1 through May 16, 1963.

$$T_0 = 38156.0 \text{ MJD}$$

$$\omega = (26^\circ 901 \pm 8) + (4^\circ 899 \pm 2)t - .000487t^2 + .2484 \cos \omega$$

$$\Omega = (92^\circ 911 \pm 3) - (3^\circ 7457 \pm 7)t - .000325t^2 + .0066 \cos \omega$$

$$i = (38^\circ 905 \pm 1) + .000342t - .0045 \sin \omega$$

$$e = (.12587 \pm 4) + (.48 \pm 7) \times 10^{-4}t + .17 \times 10^{-6}t^2 + .0005258 \sin \omega$$

$$M = (.89772 \pm 3) + (12.304044 \pm 5)t + (.0003976 \pm 4)t^2 \\ + (.580 \pm 8) \times 10^{-5}t^3 - (.148 \pm 9) \times 10^{-6}t^4 - .0006974 \cos \omega$$

Standard error of one observation: $\sigma = \pm 2.55$.

The following elements are based on 119 observations and are valid for the period May 16 through June 1, 1963.

$$T_0 = 38173.0 \text{ MJD}$$

$$\omega = (110^\circ 241 \pm 6) + (4^\circ 912 \pm 1)t - .000487t^2 + .2484 \cos \omega$$

$$\Omega = (29^\circ 126 \pm 4) - (3^\circ 7586 \pm 5)t - .000325t^2 + .0066 \cos \omega$$

$$i = (38^\circ 907 \pm 2) + .000342t - .0045 \sin \omega$$

$$e = (.12681 \pm 3) + (.65 \pm 5) \times 10^{-4}t + .17 \times 10^{-6}t^2 + .0005258 \sin \omega$$

$$M = (.18781 \pm 2) + (12.317370 \pm 3)t + (.0002992 \pm 4)t^2 \\ - (.46 \pm 39) \times 10^{-7}t^3 + (.432 \pm 7) \times 10^{-6}t^4 - .0006974 \cos \omega$$

Standard error of one observation: $\sigma = \pm 2.20$.

The following elements are based on 49 observations and are valid for the period June 1 through June 15, 1963.

$$T_0 = 38187.0 \text{ MJD}$$

$$\omega = (178^\circ 902 \pm 7) + (4^\circ 903 \pm 2)t - .000487t^2 + .2484 \cos \omega$$

$$\Omega = (336^\circ 459 \pm 3) - (3^\circ 767 \pm 1)t - .000325t^2 + .0066 \cos \omega$$

$$i = (38^\circ 907 \pm 3) + .000342t - .0045 \sin \omega$$

$$e = (.12767 \pm 4) + (.49 \pm 13) \times 10^{-4}t + .17 \times 10^{-6}t^2 + .0005258 \sin \omega$$

$$M = (.70006 \pm 2) + (12.327689 \pm 9)t + (.000428 \pm 1)t^2 \\ + (.196 \pm 8) \times 10^{-4}t^3 + (.21 \pm 1) \times 10^{-5}t^4 - .0006974 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.65$.

The following elements are based on 147 observations and are valid for the period June 15 through July 1, 1963.

$$T_0 = 38203.0 \text{ MJD}$$

$$\omega = (257^\circ 563 \pm 9) + (4^\circ 910 \pm 2)t - .000487t^2 + .2484 \cos \omega$$

$$\Omega = (276^\circ 084 \pm 5) - (3^\circ 7786 \pm 9)t - .000325t^2 + .0066 \cos \omega$$

$$i = (38^\circ 911 \pm 1) + .000342t - .0045 \sin \omega$$

$$e = (.12884 \pm 4) + (.77 \pm 8) \times 10^{-4}t + .17 \times 10^{-6}t^2 + .0005258 \sin \omega$$

$$M = (.05675 \pm 3) + (12.340895 \pm 5)t + (.0002810 \pm 8)t^2$$

$$- (.419 \pm 5) \times 10^{-5}t^3 + (.49 \pm 1) \times 10^{-6}t^4 - .0006974 \cos \omega$$

Standard error of one observation: $\sigma = \pm 2.88$.

II. SAO mean elements -- Satellite 1961 Delta 1

1-30 April 1963

T (MJD)	ω	Ω	i	e	M	n	$n'/2$	q	N	D	ζ
38120.0	211.163 6	227.476 4	38.889 2	.12318 3	*30342 1	12.28501 1	-252E-3 2	6.955612	32 4	-37	
38121.0	216.064 6	223.744 4	38.886 2	.12327 3	*58863 2	12.28551 1	-248E-3 2	6.954739	31 4	-38	
38122.C	220.017 4	38.889 2	.12327 3	*87439 2	12.28606 1	-238E-3 2	6.954557	35 4	-41		
38123.0	225.832 7	216.278 4	38.889 2	.12327 3	*16064 2	12.28654 2	-239E-3 3	6.954337	34 4	-42	
38124.0	230.719 8	212.550 5	38.888 2	.12335 4	*44735 2	12.28707 1	-227E-3 2	6.953472	36 4	-47	
38125.0	235.605 9	208.818 6	38.892 3	.12337 4	*73463 2	12.28758 1	-281E-3 2	6.953174	30 4	-44	
38126.0	240.502 8	205.087 5	38.893 2	.12348 6	*02245 2	12.28816 2	-273E-3 4	6.952055	21 4	-40	
38127.0	245.377 7	201.363 5	38.896 2	.12348 4	*31087 2	12.28870 1	-259E-3 2	6.951871	24 4	-41	
38128.0	250.276 7	197.625 5	38.895 1	.12358 4	*59975 2	12.28923 1	-253E-3 3	6.950839	28 4	-41	
38129.C	255.160 6	193.898 4	38.895 1	.12364 3	*88917 1	12.28970 1	-243E-3 2	6.950244	30 4	-39	
38130.0	260.052 5	190.162 4	38.8935 9	.12377 2	*179062 9	12.290165 9	-239E-3 2	6.948999	34 4	-34	
38131.0	264.940 5	186.429 4	38.8949 9	.12385 2	*46944 1	12.290703 9	-257E-3 2	6.948147	35 4	-38	
38132.C	265.834 5	182.691 4	38.894 1	.12397 3	*76032 1	12.29124 1	-282E-3 4	6.947044	36 4	-43	
38133.0	274.710 6	178.958 4	38.895 1	.12401 3	*05180 1	12.29184 1	-312E-3 3	6.946478	42 4	-46	
38134.0	279.597 5	175.219 3	38.894 1	.12401 2	*34389 1	12.292457 8	-307E-3 2	6.946233	45 4	-43	
38135.0	284.481 5	171.483 3	38.895 1	.12409 2	*63661 1	12.293008 9	-279E-3 2	6.945435	56 4	-56	
38136.0	289.374 4	167.741 2	38.8971 9	.12419 1	*929866 7	12.293578 7	-251E-3 1	6.944355	48 4	-36	
38137.0	294.264 5	164.006 3	38.899 1	.12431 2	*223623 9	12.294075 9	-254E-3 2	6.943264	52 4	-41	
38138.0	299.147 5	160.268 3	38.900 1	.12438 3	*51792 1	12.294604 7	-262E-3 1	6.942463	50 4	-42	
38139.0	304.040 5	156.526 3	38.900 1	.12443 3	*81272 1	12.295140 9	-260E-3 2	6.941870	44 4	-40	
38140.C	308.910 6	152.797 3	38.900 1	.12463 4	*10808 1	12.29560 1	-249E-3 2	6.940104	48 4	-46	
38141.0	313.803 5	149.057 3	38.900 1	.12468 3	*40389 1	12.29610 1	-229E-3 3	6.939523	49 4	-44	
38142.0	318.684 5	145.320 3	38.899 1	.12475 3	*70099 1	12.29658 1	-229E-3 3	6.938821	46 4	-41	
38143.0	323.558 6	141.578 4	38.902 1	.12486 3	*99699 2	12.29706 1	-224E-3 2	6.937790	46 4	-39	
38144.0	328.453 6	137.837 4	38.901 1	.12494 5	*29416 2	12.29743 2	-196E-3 2	6.937006	46 4	-39	
38145.C	333.340 5	134.093 3	38.903 1	.12510 4	*59173 1	12.297799 9	-195E-3 2	6.935578	51 4	-41	
38146.0	338.227 4	130.350 2	38.904 8	.12522 3	*88969 1	12.29817 1	-206E-3 2	6.934512	58 4	-37	
38147.0	343.117 4	126.608 2	38.9037 7	.12533 3	*18805 1	12.29859 1	-212E-3 2	6.933494	67 4	-40	
38148.0	348.002 4	122.868 2	38.9041 8	.12540 3	*48686 1	12.299001 8	-202E-3 1	6.932786	76 4	-46	
38149.0	352.886 6	119.126 3	38.904 1	.12552 3	*78605 2	12.29943 2	-212E-3 2	6.931640	55 4	-61	

MJD	ω	Ω	i	e	M	n	$n'/2$	q	N	D	σ
38150.0	357.79 1	115.382 4	38.905 2	.12560 5	.08564 3	12.29996 3	.277E-3 3	6.930789	60	4	.95
38151.0	2.673 5	111.637 2	38.9050 9	.12566 2	.38583 2	12.30054 1	.339E-3 1	6.930087	65	4	.40
38152.0	7.559 6	107.896 2	38.905 1	.12580 3	.68671 2	12.30123 2	.334E-3 2	6.928766	64	4	.43
38153.0	12.437 7	104.152 3	38.904 1	.12586 3	.98828 3	12.30191 2	.351E-3 2	6.927983	59	4	.49
38154.0	17.331 8	100.410 3	38.904 1	.12595 4	.29050 3	12.30256 2	.372E-3 3	6.927096	48	4	.50
38155.0	22.226 9	96.662 4	38.903 2	.12604 4	.59346 3	12.30330 2	.374E-3 2	6.926081	49	4	.50
38156.0	27.11 1	92.915 4	38.903 2	.12611 4	.89719 3	12.30406 2	.374E-3 3	6.925232	42	4	.50
38157.0	32.005 8	89.172 4	38.903 2	.12625 3	.20163 2	12.30489 2	.398E-3 2	6.923842	43	4	.45
38158.0	36.882 7	85.429 3	38.905 1	.12628 3	.50692 2	12.30566 2	.420E-3 3	6.923307	40	4	.34
38159.0	41.759 7	81.680 3	38.905 1	.12633 3	.81306 2	12.30653 1	.459E-3 2	6.922553	41	4	.44
38160.0	46.652 5	77.932 3	38.9040 9	.12643 2	.12008 1	12.30747 1	.469E-3 1	6.921429	39	4	.34
38161.0	51.536 6	74.184 3	38.903 1	.12653 2	.42805 2	12.30840 1	.465E-3 1	6.920269	33	4	.37
38162.0	56.424 6	70.432 3	38.903 1	.12664 2	.73695 1	12.30932 1	.469E-3 1	6.919046	33	4	.40
38163.0	61.331 8	66.686 5	38.903 1	.12675 3	.04673 2	12.31020 2	.461E-3 2	6.917892	27	4	.56
38164.0	66.210 9	62.932 6	38.903 2	.12677 4	.35753 2	12.31115 2	.426E-3 3	6.917352	31	4	.72
38165.0	71.083 6	59.174 4	38.903 1	.12684 2	.66921 1	12.31198 1	.377E-3 2	6.916488	34	4	.47
38166.0	75.956 6	55.422 3	38.902 1	.12689 2	.98162 1	12.31271 1	.367E-3 1	6.915801	40	4	.45
38167.0	80.839 6	51.667 3	38.902 1	.12695 2	.29474 1	12.31341 1	.357E-3 1	6.915063	43	4	.42
38168.0	85.717 5	47.911 3	38.902 1	.12702 2	.60857 1	12.31413 1	.360E-3 2	6.914280	36	4	.44
38169.0	90.601 7	44.155 4	38.902 3	.12708 3	.92312 1	12.31485 1	.363E-3 2	6.913485	37	4	.57
38170.0	95.490 8	40.399 5	38.903 2	.12720 4	.23836 2	12.31558 2	.352E-3 3	6.912269	31	4	.68
38171.0	100.366 8	36.641 5	38.902 2	.12721 5	.55437 2	12.31624 1	.318E-3 3	6.911949	30	4	.62
38172.0	105.254 7	32.881 5	38.902 2	.12726 4	.87097 2	12.31684 1	.302E-3 3	6.911329	30	4	.59
38173.0	110.13 1	29.132 7	38.901 3	.12734 5	.8815 2	12.31740 2	.286E-3 4	6.910477	24	4	.62
38174.0	115.02 1	25.379 9	38.900 2	.12740 6	.50589 2	12.31796 2	.275E-3 5	6.909818	23	4	.54
38175.0	119.87 2	21.617 9	38.903 3	.12751 6	.82429 4	12.31865 4	.311E-3 7	6.908653	21	4	.53
38176.0	124.76 4	17.84 1	38.906 3	.12751 7	.1433 2	12.3193 1	.312E-3 7	6.908478	21	4	.54
38177.0	129.76 2	14.08 1	38.902 3	.12760 5	.46244 7	12.32008 6	.355E-3 6	6.907442	28	4	.60
38178.0	134.569 9	10.325 5	38.902 2	.12757 4	.78295 2	12.32063 4	.402E-3 4	6.907425	33	4	.50
38179.0	139.45 1	6.566 5	38.903 3	.12764 4	.10403 3	12.32146 3	.404E-3 6	6.906620	31	4	.51
38180.0	144.341 8	2.811 5	38.901 2	.12766 3	.42587 2	12.32219 1	.3866E-3 3	6.906178	34	4	.45

T (MD)	ω	Ω	i	e	M	n	$n'/2$	q	N	D	σ
38181.0	149.37 6	359.040 7	38.902 4	.12779 9	.7480 2	12.3234 2	.400E-3 5	6.904712	28 4	.46	
38182.0	154.16 1	355.275 7	38.905 4	.12765 4	.07180 2	12.3239 4	.424E-3 5	6.905615	22 4	.48	
38183.0	159.1 1	351.514 7	38.905 5	.12777 1	.3961 4	12.3247 4	.413E-3 5	6.905124	23 4	.48	
38184.0	163.249 8	347.743 4	38.909 4	.12775 5	.72115 3	12.32540 1	.385E-3 3	6.905287	20 4	.53	
38185.0	168.36 1	343.981 4	38.911 5	.12777 7	.04696 3	12.32624 4	.334E-3 8	6.903793	20 4	.68	
38186.0	173.76 1	340.214 5	38.912 5	.12781 7	.37350 3	12.32683 3	.34E-3 1	6.903253	20 4	.78	
38187.0	178.68 2	336.440 9	38.92 1	.12777 1	.70071 5	12.32760 5	.40E-3 1	6.903832	20 4	1.42	
38188.0	183.565 9	332.683 4	38.905 5	.12780 5	.02884 2	12.32863 2	.522E-3 5	6.902649	22 4	.68	
38189.0	188.499 9	328.913 4	38.907 4	.12782 5	.35799 2	12.32975 2	.488E-3 9	6.902110	20 4	.58	
38190.0	193.420 8	325.143 4	38.908 3	.12781 3	.68815 2	12.33060 1	.456E-3 2	6.901860	25 4	.54	
38191.0	198.322 8	321.376 5	38.908 3	.12783 3	.01925 2	12.33152 2	.434E-3 2	6.901338	27 4	.53	
38192.0	203.268 6	317.605 4	38.909 2	.12785 2	.35112 1	12.33238 1	.433E-3 3	6.900858	23 4	.39	
38193.0	208.182 9	313.833 6	38.910 4	.12790 4	.68392 2	12.33324 2	.438E-3 4	6.900148	24 4	.60	
38194.0	213.103 7	310.061 5	38.909 3	.12799 3	.01759 2	12.33408 2	.434E-3 3	6.899125	23 4	.48	
38195.0	218.041 7	306.296 5	38.911 3	.12802 4	.35207 2	12.33498 1	.427E-3 3	6.898545	24 4	.50	
38196.0	222.98 1	302.517 6	38.907 4	.12808 4	.68744 2	12.33589 3	.386E-3 6	6.897730	26 4	.63	
38197.0	227.885 8	298.745 5	38.912 3	.12812 3	.02366 2	12.33665 1	.397E-3 4	6.897106	31 4	.59	
38198.0	232.818 4	294.966 3	38.912 2	.12817 2	.360604 8	12.33745 1	.432E-3 3	6.896457	25 4	.29	
38199.0	237.753 7	291.190 5	38.914 3	.12818 3	.69839 2	12.33825 2	.421E-3 4	6.896057	31 4	.50	
38200.0	242.664 6	287.412 5	38.909 3	.12831 3	.03711 2	12.33912 1	.379E-3 4	6.894722	37 4	.51	
38201.0	247.605 6	283.639 5	38.914 3	.12822 3	.37651 2	12.33991 3	.322E-3 3	6.895109	40 4	.41	
38202.0	252.538 6	279.859 5	38.912 3	.12829 3	.71656 2	12.34040 2	.284E-3 3	6.894420	41 4	.44	
38203.0	257.446 5	276.083 5	38.913 2	.12837 3	.05723 2	12.340925 9	.252E-3 2	6.893540	39 4	.36	
38204.0	262.365 8	272.309 6	38.915 3	.12841 4	.39836 2	12.34144 2	.240E-3 4	6.893029	37 4	.42	
38205.0	267.283 9	268.530 7	38.917 3	.12846 5	.73997 3	12.34192 2	.262E-3 3	6.892439	36 4	.55	
38206.0	272.216 7	264.742 6	38.913 2	.12862 3	.08204 2	12.34242 2	.323E-3 3	6.891017	35 4	.42	
38207.0	277.168 7	260.967 5	38.916 2	.12867 3	.42468 2	12.34312 2	.317E-3 3	6.890469	36 4	.40	
38208.0	282.112 8	257.183 6	38.915 2	.12873 3	.76800 2	12.34366 3	.286E-3 3	6.889680	37 4	.40	
38209.0	287.10 3	253.406 5	38.9184 8	.12897 9	.1117 1	12.3440 1	.279E-3 3	6.887640	38 4	.44	
38210.0	291.99 1	249.623 3	38.9180 5	.12898 8	.45630 1	12.34492 6	.289E-3 3	6.887227	43 4	.39	

Table 7

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1961 DELTA 1

MJD	Z	φ	ψ	D. R.A.	$\dot{\varphi}$
PERIGEE IN SUNLIGHT					
38120.	579.	-19.0	66.1	63.0	-0.334E-05
38121.	579.	-21.7	66.7	62.7	-0.329E-05
38122.	580.	-24.3	67.5	62.5	-0.315E-05
38123.	580.	-26.8	68.4	62.5	-0.317E-05
38124.	580.	-29.1	69.6	62.8	-0.367E-05
38125.	581.	-31.2	70.8	63.2	-0.372E-05
38126.	580.	-33.1	72.3	63.9	-0.362E-05
38127.	580.	-34.8	73.8	64.8	-0.343E-05
38128.	580.	-36.2	75.4	65.9	-0.335E-05
38129.	580.	-37.4	77.0	67.2	-0.322E-05
38130.	579.	-38.2	78.6	68.6	-0.316E-05
38131.	578.	-38.7	80.2	70.2	-0.340E-05
38132.	577.	-38.9	81.6	71.8	-0.373E-05
38133.	577.	-38.7	83.0	73.4	-0.413E-05
38134.	576.	-38.3	84.3	75.0	-0.406E-05
38135.	575.	-37.4	85.4	76.4	-0.369E-05
38136.	573.	-36.3	86.3	77.7	-0.332E-05
38137.	572.	-34.9	87.0	78.8	-0.336E-05
38138.	571.	-33.3	87.5	79.7	-0.347E-05
38139.	569.	-31.4	87.7	80.3	-0.344E-05
38140.	567.	-29.3	87.7	80.8	-0.329E-05
38141.	566.	-27.0	87.5	81.0	-0.303E-05

Table 7 (cont.)

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1961 DELTA 1

MJD	Z	φ	ψ	D. R.A.	\dot{P}
38142.	564.	-24.5	87.0	81.0	-0.303E-05
38143.	562.	-21.9	86.3	80.8	-0.296E-05
38144.	561.	-19.2	85.5	80.5	-0.259E-05
38145.	559.	-16.4	84.4	80.0	-0.258E-05
38146.	557.	-13.5	83.1	79.4	-0.272E-05
38147.	556.	-10.5	81.7	78.7	-0.280E-05
38148.	555.	-7.5	80.2	77.9	-0.267E-05
38149.	553.	-4.5	78.6	77.0	-0.280E-05
38150.	552.	-1.4	77.0	76.1	-0.366E-05
38151.	552.	1.7	75.3	75.2	-0.448E-05
38152.	551.	4.7	73.7	74.4	-0.441E-05
38153.	550.	7.8	72.1	73.5	-0.464E-05
38154.	549.	10.8	70.6	72.7	-0.492E-05
38155.	549.	13.7	69.2	72.0	-0.494E-05
38156.	549.	16.6	68.0	71.4	-0.494E-05
38157.	548.	19.4	66.9	70.9	-0.526E-05
38158.	548.	22.1	66.0	70.5	-0.555E-05
38159.	548.	24.7	65.3	70.3	-0.606E-05
38160.	548.	27.2	64.9	70.3	-0.619E-05
38161.	547.	29.5	64.6	70.5	-0.614E-05
38162.	547.	31.5	64.6	70.9	-0.619E-05
38163.	546.	33.4	64.8	71.5	-0.608E-05
38164.	546.	35.1	65.2	72.3	-0.562E-05
38165.	546.	36.4	65.8	73.3	-0.497E-05
38166.	545.	37.5	66.5	74.5	-0.484E-05

Table 7 (cont.)

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1961 DELTA 1

MJD	Z	φ	ψ	D. R.A.	\dot{P}
38167.	545.	38.3	67.3	75.9	-0.471E-05
38168.	544.	38.8	68.3	77.4	-0.475E-05
38169.	544.	38.9	69.3	78.9	-0.479E-05
38170.	542.	38.7	70.4	80.4	-0.464E-05
38171.	542.	38.2	71.4	81.8	-0.419E-05
38172.	541.	37.3	72.5	83.2	-0.398E-05
38173.	540.	36.1	73.6	84.3	-0.377E-05
38174.	538.	34.7	74.6	85.3	-0.362E-05
38175.	537.	33.0	75.5	86.0	-0.410E-05
38176.	536.	31.1	76.4	86.5	-0.411E-05
38177.	534.	28.9	77.3	86.9	-0.468E-05
38178.	533.	26.6	78.0	86.9	-0.530E-05
38179.	532.	24.1	78.6	86.8	-0.532E-05
38180.	531.	21.5	79.1	86.5	-0.508E-05
38181.	529.	18.7	79.7	86.1	-0.527E-05
38182.	529.	15.9	80.0	85.4	-0.558E-05
38183.	528.	12.9	80.4	84.8	-0.544E-05
38184.	527.	10.0	80.6	83.9	-0.507E-05
38185.	526.	7.0	80.9	83.0	-0.440E-05
38186.	525.	3.9	81.2	82.1	-0.448E-05
38187.	525.	0.8	81.5	81.1	-0.526E-05
38188.	524.	-2.2	81.8	80.1	-0.687E-05
38189.	524.	-5.3	82.2	79.2	-0.642E-05
38190.	524.	-8.4	82.6	78.2	-0.600E-05
38191.	524.	-11.4	83.1	77.4	-0.571E-05
38192.	524.	-14.4	83.7	76.6	-0.569E-05

Table 7 (cont.)

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1961 Delta 1

MJD	Z	φ	ψ	D. R.A.	\dot{P}
38193.	524.	-17.3	84.5	75.9	-0.576E-05
38194.	523.	-20.1	85.3	75.4	-0.571E-05
38195.	523.	-22.8	86.2	75.0	-0.561E-05
38196.	523.	-25.4	87.3	74.8	-0.507E-05
38197.	523.	-27.8	88.4	74.8	-0.522E-05
38198.	523.	-30.0	89.6	75.0	-0.568E-05
38199.	524.	-32.1	90.9	75.4	-0.553E-05
38200.	523.	-33.9	92.2	76.0	-0.498E-05
38201.	524.	-35.5	93.5	76.9	-0.423E-05
38202.	524.	-36.8	94.9	78.0	-0.373E-05
38203.	523.	-37.8	96.2	79.2	-0.331E-05
38204.	523.	-38.5	97.5	80.6	-0.315E-05
38205.	523.	-38.9	98.7	82.0	-0.344E-05
38206.	521.	-38.9	99.7	83.6	-0.424E-05
38207.	520.	-38.6	100.7	85.1	-0.416E-05
38208.	519.	-37.9	101.5	86.5	-0.375E-05
38209.	517.	-36.9	102.1	87.8	-0.366E-05
38210.	516.	-35.6	102.4	88.9	-0.379E-05

Satellite 1962 Alpha Epsilon 1 (Telstar 1)

Maria Gutierrez

I. SAO smoothed elements

The following elements are based on 122 observations and are valid for the period April 1 through May 1, 1963.

$$T_0 = 38134.0 \text{ MJD}$$

$$\omega = (358^\circ 371 \pm 6) + (1^\circ 9871 \pm 6)t - .62 \times 10^{-5}t^2 + .1142 \cos \omega$$

$$\Omega = (45^\circ 699 \pm 2) - (1^\circ 8588 \pm 2)t - .71 \times 10^{-6}t^2 + .0145 \cos \omega$$

$$i = (44^\circ 808 \pm 1) - .83 \times 10^{-4}t - .0077 \sin \omega$$

$$e = (.24240 \pm 1) + .87 \times 10^{-6}t + .0005183 \sin \omega$$

$$M = (.82816 \pm 2) + (9.126180 \pm 2)t + (.47 \pm 38) \times 10^{-7}t^2 \\ + (.77 \pm 44) \times 10^{-8}t^3 - .0003171 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1^\circ 03$.

The following elements are based on 160 observations and are valid for the period May 1 through June 1, 1963.

$$T_0 = 38164.0 \text{ MJD}$$

$$\omega = (57^\circ 995 \pm 5) + (1^\circ 9867 \pm 5)t - .62 \times 10^{-5}t^2 + .1142 \cos \omega$$

$$\Omega = (349^\circ 928 \pm 2) - (1^\circ 8592 \pm 2)t - .71 \times 10^{-6}t^2 + .0145 \cos \omega$$

$$i = (44^\circ 8054 \pm 9) - .83 \times 10^{-4}t - .0077 \sin \omega$$

$$e = (.242450 \pm 9) + .87 \times 10^{-6}t + .0005183 \sin \omega$$

$$M = (.61364 \pm 2) + (9.126186 \pm 2)t - (.71 \pm 48) \times 10^{-7}t^2 \\ + (.10 \pm 4) \times 10^{-7}t^3 - .0003171 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1^\circ 00$.

The following elements are based on 230 observations and are valid for the period June 1 through July 1, 1963.

$$T_0 = 38196.0 \text{ MJD}$$

$$\omega = (121^{\circ}590 \pm 5) + (1^{\circ}9858 \pm 5)t - .62 \times 10^{-5}t^2 + .1142 \cos \omega$$

$$\Omega = (290^{\circ}430 \pm 1) - (1.8588 \pm 2)t - .71 \times 10^{-6}t^2 + .0145 \cos \omega$$

$$i = (44^{\circ}803 \pm 1) - .83 \times 10^{-4}t - .0077 \sin \omega$$

$$e = (.24233 \pm 1) + .87 \times 10^{-6}t + .0005183 \sin \omega$$

$$M = (.65156 \pm 2) + (9.126188 \pm 2)t - (.24 \pm 3) \times 10^{-6}t^2 \\ - (.15 \pm 4) \times 10^{-7}t^3 - .0003171 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.08$.

III. SAO mean elements -- Satellite 1962 Alpha Epsilon 1

3 April - 30 June 1963

(MJD)	Ω	ω	i	e	M	n	$n'/2$	q	\dot{n}	\dot{D}	σ
381122.0	334.64 1	68.020 5	44.806 2	.24210 3	.31368 5	9.126187 6	-3E-6 9	7.330485	.28	6	.40
381126.0	342.62 5	60.579 9	44.809 5	.24229 8	.8183 2	9.12621 1	-1E-5 6	7.328572	.9	6	.42
381130.0	350.4 1	53.16 2	44.810 4	.2425 2	.3234 3	9.12617 2	.4E-5 5	7.326642	10	6	.42
381134.0	358.44 4	45.717 5	44.809 3	.2422 1	.8280 2	9.126173 3	.2E-5 2	7.329549	20	6	.34
381138.0	6.45 1	38.277 3	44.808 1	.24250 2	.33247 5	9.126186 2	.3E-5 1	7.326567	37	6	.33
381142.0	14.39 1	30.842 3	44.811 2	.24259 2	.83726 4	9.126192 2	-1E-5 1	7.325663	33	6	.33
381146.0	22.32 2	23.408 3	44.805 2	.24261 5	.34205 8	9.126184 4	.2E-5 2	7.325564	25	6	.42
381150.0	30.29 2	15.968 4	44.803 2	.24265 3	.84672 6	9.126179 5	.4E-6 9	7.325136	21	6	.40
381154.0	38.21 2	8.537 8	44.804 2	.24264 7	.35148 6	9.126179 6	.3E-5 4	7.325274	21	6	.43
381158.0	46.13 3	1.097 8	44.804 .4	.24281 5	.85636 9	9.126194 9	-2E-6 9	7.323570	14	6	.60
381162.0	54.13 2	353.652 5	44.804 3	.24287 3	.3609 1	9.126191 5	-1E-5 4	7.322963	15	6	.40
381166.0	62.026 8	346.214 2	44.801 1	.24294 1	.86586 3	9.126190 2	-2E-5 1	7.322357	42	6	.29
381170.0	69.941 8	338.778 3	44.794 2	.24298 2	.37069 3	9.126191 3	.2E-5 2	7.321964	45	6	.42
381174.0	77.90 1	331.343 4	44.793 2	.24296 3	.87536 5	9.126194 4	.1E-5 2	7.322121	30	6	.46
381178.0	85.827 8	323.900 2	44.794 2	.24294 2	.38019 3	9.126191 3	.2E-5 2	7.322344	48	6	.41
381182.0	93.75 1	316.463 3	44.791 2	.24295 4	.88504 5	9.126194 3	-3E-5 2	7.322234	31	6	.41
381186.0	101.69 2	309.013 4	44.791 4	.24296 4	.38983 7	9.126193 4	-3E-5 3	7.322087	16	6	.44
381190.0	109.64 2	301.585 7	44.795 4	.24283 5	.89450 9	9.126181 7	-2E-5 7	7.323419	11	6	.48
381194.0	117.55 1	294.144 3	44.797 2	.24281 4	.39936 5	9.126191 4	-2E-5 2	7.323565	37	6	.44
381198.0	125.494 7	286.704 2	44.799 2	.24278 2	.90413 2	9.126195 2	-1E-5 1	7.323901	76	6	.41
382002.0	133.435 6	279.265 2	44.799 2	.24271 2	.40888 2	9.126180 2	-1.6E-5 8	7.324580	86	6	.43
382006.0	141.375 6	271.830 2	44.795 1	.24256 2	.91356 2	9.126176 1	.6E-6 8	7.325960	61	6	.31
38210.0	149.31 1	264.398 3	44.798 2	.24250 3	.41829 4	9.126188 2	.3E-5 1	7.326626	36	6	.32

Table 8

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1962 ALPHA EPSILON 1

MJD	Z	ϕ	ψ	D. R.A.	\dot{P}
PERIGEE IN SUNLIGHT					
38122.	954.	-17.6	43.7	37.9	0.720E-08
38126.	951.	-12.2	37.6	32.9	0.240E-07
38130.	949.	-6.7	31.1	27.5	-0.961E-07
38134.	951.	-1.1	24.4	22.1	-0.480E-07
38138.	948.	4.5	17.6	16.6	-0.720E-07
38142.	948.	10.1	11.2	11.2	0.240E-07
38146.	949.	15.5	6.1	5.9	-0.480E-07
38150.	949.	20.8	6.1	1.0	0.961E-08
38154.	951.	25.8	10.4	356.4	-0.720E-07
38158.	951.	30.5	15.1	352.3	0.480E-08
38162.	952.	34.8	19.3	349.0	0.240E-07
38166.	952.	38.5	22.7	346.3	0.480E-07
38170.	953.	41.4	25.2	344.5	-0.480E-07
38174.	954.	43.5	26.6	343.5	-0.240E-07
38178.	955.	44.6	27.1	342.9	-0.480E-07
38182.	954.	44.7	26.8	342.6	0.720E-07
38186.	954.	43.6	25.8	342.0	0.720E-07
38190.	954.	41.6	24.6	340.9	0.480E-07
38194.	954.	38.7	23.7	338.9	0.480E-07
38198.	953.	35.0	23.8	336.1	0.240E-07
38202.	952.	30.8	25.4	332.5	0.384E-07
38206.	952.	26.1	28.9	328.2	-0.144E-07
38210.	951.	21.1	33.9	323.4	-0.720E-07

Satellite 1962 Beta Mu 1 (Anna 1B)

Beatrice Miller

I. SAO smoothed elements

The following elements are based on 257 observations and are valid for the period April 1 through May 1, 1963.

$$T_0 = 38135.0 \text{ MJD}$$

$$\omega = (341^\circ 61 \pm 6) + (2^\circ 916 \pm 6)t + 6^\circ 2697 \cos \omega$$

$$\Omega = (172^\circ 8677 \pm 6) - (3^\circ 60899 \pm 7)t + 0^\circ 00085 \cos \omega$$

$$i = (50^\circ 1416 \pm 5) - 0^\circ 0002 \sin \omega$$

$$e = (.007046 \pm 5) + (.88 \pm 69) \times 10^{-6}t + .0007540 \sin \omega$$

$$M = (.5787 \pm 2) + (13.34498 \pm 2)t + (.175 \pm 1) \times 10^{-5}t^2 \\ + (.77 \pm 15) \times 10^{-8}t^3 - .015988 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1^\circ 18$.

The following elements are based on 172 observations and are valid for the period May 1 through June 1, 1963.

$$T_0 = 38164.0 \text{ MJD}$$

$$\omega = (67^\circ 07 \pm 5) + (2^\circ 965 \pm 5)t + 6^\circ 2697 \cos \omega$$

$$\Omega = (68^\circ 209 \pm 1) - (3^\circ 60882 \pm 9)t + 0^\circ 00085 \cos \omega$$

$$i = (50^\circ 1406 \pm 8) - 0^\circ 0002 \sin \omega$$

$$e = (.00707 \pm 1) + (.01 \pm 11) \times 10^{-5}t + .0007540 \sin \omega$$

$$M = (.5321 \pm 1) + (13.34494 \pm 1)t + (.76 \pm 2) \times 10^{-6}t^2 \\ - (.22 \pm 2) \times 10^{-7}t^3 - .015988 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1^\circ 25$.

The following elements are based on 133 observations and are valid for the period June 1 through July 1, 1963.

$$T_0 = 38195.0 \text{ MJD}$$

$$\omega = (159^\circ.15 \pm 9) + (2^\circ.952 \pm 9)t + 6^\circ.2697 \cos \omega$$

$$\Omega = (316^\circ.327 \pm 1) - (3^\circ.6093 \pm 1)t + ^\circ.00085 \cos \omega$$

$$i = (50^\circ.140 \pm 1) - ^\circ.0002 \sin \omega$$

$$e = (.007087 \pm 8) + (.157 \pm 97)t + .0007540 \sin \omega$$

$$M = (.2746 \pm 2) + (13.34494 \pm 3)t - (.159 \pm 2) \times 10^{-5}t^2 \\ - (.45 \pm 25) \times 10^{-8}t^3 - .015988 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.15$.

II. SAO mean elements -- Satellite 1962 Beta Mu 1

3 April - 30 June 1963

T (MTD)	ω	Ω	i	e	M	n	$n'/2$	q	N	D	σ
38122.0	307.0 2	219.783 2	50.141 2	.00639 2	.0859 7	13.345365 3	.2E-5 3	7.459626	27	4	.66
38126.0	319.6 4	205.351 3	50.140 2	.00654 3	.4632 1	13.345371 4	-.4E-5 4	7.458464	18	4	.71
3813C.0	332.7 3	190.909 3	50.141 2	.00665 3	.8391 7	13.345372 7	.10E-6 6	7.457690	17	4	.51
38134.0	344.5 2	176.482 3	50.142 2	.00679 1	.2190 6	13.345392 3	.2E-5 2	7.456599	22	4	.35
38138.0	356.5 1	162.043 1	50.141 1	.006937 9	.5979 4	13.345399 2	-.3E-5 2	7.455486	52	4	.47
38142.0	8.5 2	147.605 2	50.138 2	.00710 2	.9770 5	13.345409 2	.5E-5 2	7.454269	46	4	.59
38146.0	19.8 1	133.167 2	50.138 1	.00723 2	.3582 3	13.345430 2	.1E-5 2	7.453276	51	4	.43
38150.0	31.1 1	118.729 2	50.145 1	.00729 3	.7392 3	13.345445 2	.1E-5 2	7.452828	44	4	.52
38154.0	42.6 1	104.300 2	50.137 2	.00759 3	.1196 3	13.345469 4	.7E-5 3	7.450576	25	4	.51
38158.0	53.3 1	89.860 4	50.135 3	.00769 6	.5022 4	13.345460 4	-.4E-5 3	7.449792	13	4	.43
38162.0	63.93 9	75.422 3	50.142 1	.00770 3	.8852 3	13.345482 3	.1E-5 3	7.449745	24	4	.33
38166.0	74.6 1	60.996 5	50.141 2	.00773 4	.2678 3	13.345478 3	.2E-5 2	7.449515	32	4	.48
38170.0	85.5 2	46.554 5	50.141 3	.00792 4	.6500 5	13.345486 4	.8E-5 4	7.448105	24	4	.65
38174.0	96.0 2	32.13 1	50.14 1	.00779 5	.0333 6	13.345487 4	-.7E-5 4	7.449085	17	4	.48
38178.0	106.2 4	17.69 1	50.127 9	.00779 6	.418 1	13.345515 9	.7E-5 7	7.449063	14	4	.73
38182.0	117.4 5	3.239 8	50.155 9	.00777 8	.799 1	13.345503 9	.7E-5 6	7.449230	11	4	.87
38186.0	128.4 2	348.809 2	50.141 2	.00761 3	.1806 6	13.345494 4	-.7E-5 3	7.450401	17	4	.51
38190.0	138.9 2	334.372 2	50.142 3	.00753 2	.5640 6	13.345468 3	-.5E-5 2	7.450979	14	4	.43
38194.0	15C.3 2	319.930 3	50.136 3	.00739 2	.9447. 5	13.345464 2	.2E-5 3	7.452064	14	4	.33
38198.0	162.0 3	305.483 4	50.125 6	.00727 2	.3248 9	13.345465 3	.4E-8 3	7.452948	22	4	.50
38202.0	173.4 3	291.056 5	50.138 4	.00709 2	.7056 7	13.345450 2	-.6E-5 2	7.454321	23	4	.41
38206.0	185.7 3	276.626 7	50.139 4	.00695 2	.0837 8	13.345451 3	.3E-5 3	7.455356	18	4	.43
38210.0	197.1 5	262.18 1	50.138 6	.00684 4	.464 1	13.345434 5	.7E-5 4	7.456199	21	4	.61

Table 9

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1962 BETA MU 1

MJD	Z	φ	ψ	D. R.A.	\dot{P}
PERIGEE IN EARTH SHADOW					
38122.	1089.	-37.8	145.3	167.9	-0.225E-07
38126.	1085.	-29.8	150.9	161.6	0.449E-07
38130.	1082.	-20.6	151.7	153.8	-0.112E-08
38134.	1079.	-11.8	144.4	143.9	-0.225E-07
38138.	1077.	-2.7	133.2	133.6	0.337E-07
PERIGEE IN SUNLIGHT					
38142.	1076.	6.5	120.4	123.1	-0.561E-07
38146.	1076.	15.1	107.3	112.4	-0.112E-07
38150.	1078.	23.4	95.1	102.3	-0.112E-07
38154.	1078.	31.3	84.6	93.5	-0.786E-07
38158.	1080.	38.0	76.0	85.3	0.449E-07
38162.	1082.	43.6	69.7	79.0	-0.112E-07
38166.	1083.	47.7	65.8	74.7	-0.225E-07
38170.	1082.	49.9	63.7	72.5	-0.898E-07
38174.	1083.	49.8	61.7	70.4	0.786E-07
38178.	1082.	47.5	58.9	67.0	-0.786E-07
38182.	1081.	43.0	55.7	63.0	-0.786E-07
38186.	1080.	37.0	50.4	56.6	0.786E-07
38190.	1078.	30.3	43.0	47.7	0.561E-07
38194.	1077.	22.4	35.2	38.3	-0.225E-07
38198.	1076.	13.7	28.2	28.0	-0.449E-10
38202.	1076.	5.1	24.6	16.9	0.674E-07
38206.	1077.	-4.4	28.4	6.3	-0.337E-07
38210.	1079.	-13.0	36.6	355.2	-0.786E-07

Satellite 1962 Beta Tau 2 (Injun 3)

Beatrice Miller

I. SAO smoothed elements

The following elements are based on 74 observations and are valid for the period April 1 through April 15, 1963.

$$T_0 = 38128.0 \text{ MJD}$$

$$\omega = (21^{\circ}68 \pm 2) - (1^{\circ}103 \pm 4)t - .0001143t^2 + .2309 \cos \omega$$

$$\Omega = (20^{\circ}486 \pm 3) - (1^{\circ}6932 \pm 8)t - .917 \times 10^{-4}t^2 + .0330 \cos \omega$$

$$i = (70^{\circ}365 \pm 4) - .0024 \sin \omega$$

$$e = (.15818 \pm 3) - (.47 \pm 8) \times 10^{-4}t + .284 \times 10^{-6}t^2 + .0007046 \sin \omega$$

$$M = (.55007 \pm 7) + (12.45946 \pm 1)t + (.000570 \pm 1)t^2$$

$$- (.59 \pm 12) \times 10^{-6}t^3 - (.13 \pm 2) \times 10^{-6}t^4 - .0006173 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.95$.

The following elements are based on 63 observations and are valid for the period April 15 through May 1, 1963.

$$T_0 = 38143.0 \text{ MJD}$$

$$\omega = (5^{\circ}1 \pm 1) - (1^{\circ}10 \pm 2)t - .0001143t^2 + .2309 \cos \omega$$

$$\Omega = (355^{\circ}021 \pm 7) - (1^{\circ}698 \pm 2)t - .917 \times 10^{-4}t^2 + .0330 \cos \omega$$

$$i = (70^{\circ}40 \pm 2) - .0024 \sin \omega$$

$$e = (.1573 \pm 2) - (.67 \pm 57) \times 10^{-4}t + .284 \times 10^{-6}t^2 + .0007046 \sin \omega$$

$$M = (.5708 \pm 5) + (12.47618 \pm 6)t + (.000462 \pm 2)t^2$$

$$- (.62 \pm 2) \times 10^{-5}t^3 + (.01 \pm 26) \times 10^{-7}t^4 - .0006173 \cos \omega$$

Standard error of one observation: $\sigma = \pm 2.63$.

The following elements are based on 32 observations and are valid for the period May 1 through May 16, 1963.

$$T_0 = 38158.0 \text{ MJD}$$

$$\omega = (348^{\circ}63 \pm 4) - (1^{\circ}090 \pm 9)t - .0001143t^2 + .2309 \cos \omega$$

$$\Omega = (329^{\circ}512 \pm 2) - (1^{\circ}7037 \pm 6)t - .917 \times 10^{-4}t^2 + .0330 \cos \omega$$

$$i = (70^{\circ}356 \pm 4) - .0024 \sin \omega$$

$$e = (.1571 \pm 2) - (.38 \pm 30) \times 10^{-4}t + .284 \times 10^{-6}t^2 + .0007046 \sin \omega$$

$$M = (.8090 \pm 2) + (12.48904 \pm 4)t + (.000426 \pm 2)t^2$$

$$- (.19 \pm 15) \times 10^{-6}t^3 + (.48 \pm 31) \times 10^{-7}t^4 - .0006173 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1^{\circ}35$.

The following elements are based on 32 observations and are valid for the period May 16 through June 1, 1963.

$$T_0 = 38173.0 \text{ MJD}$$

$$\omega = (332^{\circ}02 \pm 2) - (1^{\circ}094 \pm 6)t - .0001143t^2 + .2309 \cos \omega$$

$$\Omega = (303^{\circ}95 \pm 1) - (1^{\circ}7061 \pm 7)t - .917 \times 10^{-4}t^2 + .0330 \cos \omega$$

$$i = (70^{\circ}35 \pm 2) - .0024 \sin \omega$$

$$e = (.15659 \pm 6) - (.89 \pm 28) \times 10^{-4}t + .284 \times 10^{-6}t^2 + .0007046 \sin \omega$$

$$M = (.23339 \pm 6) + (12.49960 \pm 2)t + (.000242 \pm 2)t^2$$

$$- (.19 \pm 1) \times 10^{-5}t^3 + (.23 \pm 2) \times 10^{-6}t^4 - .0006173 \cos \omega$$

Standard error of one observation: $\sigma = \pm 2^{\circ}25$.

The following elements are based on 51 observations and are valid for the period June 1 through June 15, 1963.

$$T_0 = 38188.0 \text{ MJD}$$

$$\omega = (315^{\circ}38 \pm 8) - (1^{\circ}10 \pm 1)t - .0001143 + .2309 \cos \omega$$

$$\Omega = (278^{\circ}36 \pm 1) - (1^{\circ}7071 \pm 6)t - .917 \times 10^{-4}t^2 + .0330 \cos \omega$$

$$i = (70^{\circ}37 \pm 2) - .0024 \sin \omega$$

$$e = (.1551 \pm 4) - (.73 \pm 49) \times 10^{-4}t + .284 \times 10^{-6}t^2 + .0007046 \sin \omega$$

$$M = (.78215 \pm 9) + (12.50678 \pm 2)t + (.0002381 \pm 9)t^2$$

$$+ (.15 \pm 7) \times 10^{-6}t^3 - (.19 \pm 2) \times 10^{-6}t^4 - .0006173 \cos \omega$$

Standard error of one observation: $\sigma = \pm 2^{\circ}43$.

The following elements are based on 164 observations and are valid for the period June 15 through July 1, 1963.

$$T_0 = 38203.0 \text{ MJD}$$

$$\omega = (298^{\circ}79 \pm 2) - (1^{\circ}112 \pm 3)t - 0.0001143t^2 + 0.2309 \cos \omega$$

$$\Omega = (252^{\circ}727 \pm 8) - (1^{\circ}7089 \pm 4)t - 0.917 \times 10^{-4}t^2 + 0.0330 \cos \omega$$

$$i = (70^{\circ}366 \pm 9) - 0.0024 \sin \omega$$

$$e = (.1558 \pm 3) - (.21 \pm 43) \times 10^{-4}t + .284 \times 10^{-6}t^2 + .0007046 \cos \omega$$

$$M = (.43462 \pm 6) + (12.51336 \pm 1)t + (.0001753 \pm 5)t^2$$

$$- (.186 \pm 4) \times 10^{-5}t^3 + (.263 \pm 8) \times 10^{-6}t^4 - .0006173 \cos \omega$$

Standard error of one observation: $\sigma = \pm 2.85$.

$\frac{T}{(MJD)}$	Ψ	Ω	i	e	M	n	$n^{1/2}$	q	N	D	σ
38120.0	3C.72 1	34.053 4	70.360 4	.15889 3	.90999 5	12.450585 3	.507E-3 1	6.614859	79 6	.20	
38122.0	28.53 1	30.667 4	70.363 4	.15878 3	.81318 5	12.452659 3	.526E-3 2	6.614968	59 6	.44	
38124.0	26.33 2	27.288 5	70.368 5	.15862 5	.72059 8	12.454883 4	.581E-3 3	6.615413	39 6	.75	
38126.0	24.14 2	23.908 5	70.367 5	.15860 4	.63264 8	12.457195 4	.577E-3 2	6.614792	27 6	.68	
38128.0	21.93 2	20.518 3	70.367 3	.15848 2	.54943 6	12.459482 3	.526E-3 2	6.614889	29 6	.54	
38130.0	19.69 1	17.137 1	70.366 2	.15835 2	.47077 4	12.461678 1	.544E-3 2	6.615162	31 6	.44	
38132.0	17.54 2	13.742 2	70.365 3	.15824 4	.39612 8	12.463919 5	.587E-3 3	6.615207	32 6	.78	
38134.0	15.30 2	10.348 1	70.366 2	.15800 3	.32648 7	12.466364 3	.626E-3 2	6.616289	35 6	.73	
38136.0	13.06 3	6.954 8	70.37 1	.15775 9	.2619 1	12.468766 4	.576E-3 2	6.617367	40 6	.70	
38138.0	1C.91 3	3.561 5	70.368 9	.1577 2	.2015 2	12.471030 3	.557E-3 1	6.616933	42 6	.47	
38140.0	9.10 6	0.175 7	70.35 1	.1594 2	.1441 3	12.473201 8	.526E-3 5	6.603185	42 6	1.20	
38142.0	7.34 5	356.181 4	70.34 1	.15932 8	.0910 2	12.475258 6	.498E-3 3	6.602733	29 6	.73	
38144.0	5.49 9	353.395 3	70.338 5	.15940 8	.0421 3	12.477077 7	.437E-3 4	6.601031	19 6	1.11	
38146.0	3.0 1	349.998 1	70.340 1	.15884 8	.9990 4	12.478675 9	.398E-3 9	6.605286	13 6	.98	
38148.0	C.7 1	346.597 1	70.340 1	.15850 5	.9584 4	12.48038 1	.441E-3 5	6.607348	10 6	.92	
38150.0	357.9 1	343.169 5	70.368 6	.1574 2	.9230 4	12.48200 3	.44E-3 1	6.615553	14 6	1.08	
38152.0	355.57 4	339.75 1	70.34 2	.1570 1	.8890 2	12.48385 2	.458E-3 7	6.617973	21 6	.58	
38154.0	353.32 2	336.355 5	70.354 7	.15684 8	.8587 1	12.485656 3	.438E-3 2	6.618549	33 6	.43	
38156.0	351.16 3	332.987 4	70.401 7	.1564 1	.8315 1	12.487367 8	.419E-3 5	6.621186	36 6	.79	
38158.0	348.91 3	329.580 5	70.397 7	.1563 1	.8080 1	12.489039 6	.420E-3 4	6.621235	29 6	.79	
38160.0	346.68 5	326.174 7	70.39 1	.15627 9	.7878 2	12.490785 7	.444E-3 5	6.621235	25 6	1.10	
38162.0	344.33 1	322.741 2	70.365 3	.15642 4	.77164 3	12.492549 7	.445E-3 4	6.619374	31 6	.77	
38164.0	342.252 5	319.3339 5	70.3656 6	.156483 8	.75816 2	12.494237 2	.361E-3 1	6.618314	40 6	.38	

T (MJD)	ω	Ω	ι	e	M	n	n'/2	q	N	D	σ
38166.0	340.039 5	315.9256 5	70.3658 7	.156365 9	.74807 2	12.495638 2	*339E-3 1	6.618742	45	6	.41
38168.0	337.8C9 5	312.5172 9	70.3657 9	.15627 1	.74073 2	12.496923 2	*313E-3 1	6.619026	38	6	.35
38170.C	335.566 6	309.100 1	70.362 1	.15618 1	.73596 1	12.498122 2	*281E-3 2	6.619294	39	6	.40
38172.0	333.384 4	305.6886 9	70.3609 9	.15609 1	.73323 1	12.499184 3	*241E-3 2	6.619632	32	6	.31
38174.C	331.139 3	302.2793 6	70.3636 6	.156009 9	.73266 1	12.5001201 9	*223E-3 1	6.619954	35	6	.24
38176.0	328.929 5	298.8715 8	70.3646 5	.15600 2	.73372 2	12.501031 2	*2494E-3 6	6.619741	33	6	.43
38178.0	326.687 3	295.4578 7	70.3646 5	.15592 2	.73683 1	12.502033 1	*2585E-3 7	6.619954	36	6	.39
38180.0	324.438 5	292.0439 9	70.364 6	.15585 2	.74200 1	12.503054 1	*2531E-3 5	6.620159	39	6	.39
38182.C	322.250 4	288.619 1	70.3625 9	.15582 1	.74904 1	12.504062 1	*2449E-3 8	6.620048	44	6	.35
38184.C	320.029 6	285.204 2	70.363 1	.15573 3	.75815 2	12.505001 2	*218E-3 1	6.620404	40	6	.40
38186.0	317.819 4	281.7924 6	70.3645 7	.15578 2	.76894 2	12.505864 1	*2129E-3 9	6.619738	42	6	.35
38188.0	315.613 2	278.3777 5	70.3621 7	.15569 1	.781455 7	12.506793 1	*255E-3 1	6.620114	50	6	.35
38190.C	313.380 2	274.9608 5	70.3642 7	.15540 2	.79614 1	12.507782 1	*2282E-3 8	6.622001	60	6	.37
36192.C	311.174 4	271.5472 8	70.3674 9	.15561 1	.81249 1	12.5086556 5	*2153E-3 4	6.620874	56	6	.28
38194.0	308.935 2	268.1259 4	70.3644 4	.15542 1	.830701 7	12.5095239 7	*2209E-3 4	6.621239	64	6	.28
38196.0	306.696 3	264.7051 5	70.3639 6	.15532 1	.850673 8	12.5103975 8	*2175E-3 5	6.621739	66	6	.38
38198.0	304.432 4	261.2881 6	70.3657 9	.155318 8	.87241 1	12.511295 1	*2318E-3 9	6.621428	90	6	.57
38200.C	302.234 7	257.8718 8	70.362 1	.155296 8	.89584 2	12.512240 1	*228E-3 1	6.621268	95	6	.63
38202.0	300.032 8	254.4496 8	70.362 1	.15516 1	.92112 2	12.513035 2	*1836E-3 8	6.622085	96	6	.67
38204.0	297.786 6	251.033 2	70.368 2	.15510 2	.94796 2	12.5136958 9	*1569E-3 6	6.622324	90	6	.42
38206.0	295.562 7	247.615 1	70.369 2	.15506 2	.97592 2	12.514390 2	*1945E-3 9	6.622337	78	6	.50
38208.C	293.325 7	244.191 1	70.365 2	.15499 2	.00549 2	12.515161 1	*1946E-3 9	6.622659	92	6	.53
38210.C	291.095 6	240.7717 8	70.363 1	.15495 2	.03657 2	12.515916 1	*1838E-3 8	6.622681	74	6	.43

Table 10

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1962 BETA TAU 2

MJD	Z	φ	ψ	D. R.A.	\dot{P}
PERIGEE IN SUNLIGHT					
38120.	241.	28.8	41.8	35.6	-0.654E-05
38122.	241.	26.7	35.6	29.5	-0.678E-05
38124.	241.	24.7	29.4	23.4	-0.749E-05
38126.	240.	22.7	23.2	17.3	-0.744E-05
38128.	239.	20.6	17.2	11.2	-0.716E-05
38130.	239.	18.5	11.7	5.2	-0.701E-05
38132.	239.	16.5	7.8	359.1	-0.756E-05
38134.	239.	14.4	8.4	353.1	-0.806E-05
38136.	240.	12.3	12.9	347.0	-0.741E-05
38138.	239.	10.3	18.6	341.0	-0.716E-05
38140.	225.	8.6	24.6	335.2	-0.676E-05
38142.	225.	6.9	30.7	329.3	-0.640E-05
38144.	223.	5.2	36.9	323.4	-0.561E-05
38146.	227.	2.8	43.5	317.3	-0.511E-05
38148.	229.	0.7	50.1	311.2	-0.566E-05
38150.	237.	-2.0	57.0	304.9	-0.565E-05
38152.	240.	-4.2	63.6	298.8	-0.588E-05
38154.	240.	-6.3	70.2	292.8	-0.562E-05
38156.	243.	-8.3	76.6	286.7	-0.537E-05
38158.	244.	-10.4	83.1	280.6	-0.539E-05
38160.	244.	-12.5	89.6	274.5	-0.569E-05
38162.	242.	-14.7	96.2	268.2	-0.570E-05
38164.	242.	-16.7	102.5	262.1	-0.463E-05
PERIGEE IN EARTH SHADOW					
38166.	243.	-18.8	108.8	255.9	-0.434E-05
38168.	243.	-20.8	115.2	249.7	-0.401E-05

Table 10 (cont.)

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1962 BETA TAU 2

MJD	Z	φ	ψ	D. R.A.	\dot{P}
38170.	244.	-22.9	121.4	243.4	-0.360E-05
38172.	245.	-25.0	127.5	237.1	-0.309E-05
38174.	246.	-27.0	133.5	230.7	-0.285E-05
38176.	246.	-29.1	139.3	224.4	-0.319E-05
38178.	247.	-31.1	144.9	217.9	-0.331E-05
38180.	248.	-33.2	150.0	211.4	-0.324E-05
38182.	249.	-35.2	154.6	204.9	-0.313E-05
38184.	250.	-37.2	158.3	198.2	-0.279E-05
38186.	250.	-39.2	160.6	191.6	-0.272E-05
38188.	251.	-41.2	161.1	184.8	-0.326E-05
38190.	254.	-43.2	159.7	178.0	-0.292E-05
38192.	253.	-45.2	156.8	171.0	-0.275E-05
38194.	254.	-47.1	152.9	164.0	-0.282E-05
38196.	256.	-49.0	148.5	156.8	-0.278E-05
38198.	256.	-51.0	143.7	149.5	-0.296E-05
38200.	257.	-52.8	138.9	142.0	-0.291E-05
38202.	258.	-54.6	134.1	134.4	-0.235E-05
38204.	259.	-56.4	129.3	126.6	-0.200E-05
38206.	259.	-58.2	124.6	118.5	-0.248E-05
38208.	260.	-59.9	120.1	110.2	-0.248E-05
38210.	261.	-61.5	115.7	101.5	-0.235E-05

I. SAO smoothed elements

The following elements are based on 87 observations and are valid for the period April 1 through May 1, 1963.

$$T_0 = 38135.0 \text{ MJD}$$

$$\omega = (326^\circ 698 \pm 6) + (1^\circ 2148 \pm 9)t + .0960 \cos \omega$$

$$\Omega = (61^\circ 152 \pm 3) - (1^\circ 2794 \pm 3)t + .0158 \cos \omega$$

$$i = (47^\circ 522 \pm 1) - .0082 \sin \omega$$

$$e = (.28492 \pm 3) - (.26 \pm 31) \times 10^{-5}t + .0005027 \sin \omega$$

$$M = (.11452 \pm 2) + (7.780966 \pm 4)t - (.81 \pm 82) \times 10^{-7}t^2 \\ - .0002583 \cos \omega$$

Standard error of one observation: $\sigma = \pm 0^\circ 98$.

The following elements are based on 118 observations and are valid for the period May 1 through June 1, 1963.

$$T_0 = 38164.0 \text{ MJD}$$

$$\omega = (1^\circ 888 \pm 6) + (1^\circ 2135 \pm 6)t + .0960 \cos \omega$$

$$\Omega = (24^\circ 048 \pm 2) - (1^\circ 2797 \pm 3)t + .0158 \cos \omega$$

$$i = (47^\circ 527 \pm 1) - .0082 \sin \omega$$

$$e = (.28479 \pm 2) - (.65 \pm 25) \times 10^{-5}t + .0005027 \sin \omega$$

$$M = (.76255 \pm 2) + (7.780959 \pm 2)t - (.32 \pm 3) \times 10^{-6}t^2 \\ - .0002583 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1^\circ 10$.

The following elements are based on 150 observations and are valid for the period June 1 through July 1, 1963.

$$T_0 = 38195.0 \text{ MJD}$$

$$\omega = (39^\circ 487 \pm 4) + (1^\circ 2115 \pm 6)t + .0960 \cos \omega$$

$$\Omega = (344^\circ 384 \pm 2) - (1^\circ 2802 \pm 2)t + .0158 \cos \omega$$

$$i = (47^\circ 518 \pm 1) - .0082 \sin \omega$$

$$e = (.28468 \pm 3) - (.53 \pm 41) \times 10^{-5}t + .0005027 \sin \omega$$

$$M = (.97206 \pm 1) + (7.780948 \pm 1)t - (.29 \pm 3) \times 10^{-6}t^2 \\ - .0002583 \cos \omega$$

Standard error of one observation: $\sigma = \pm 1.10.$

III. SAO mean elements -- Satellite 1962 Beta Upsilon 1

3 April - 30 June 1963

Ω	ω	ϵ_0	i	e	M	n	$n^{1/2}$	q	η	D	σ	
38122.C	31C.986	3	77.8201	6	47.5212	6	.284634	9	.961614	6	-1E-5	1
38126.C	315.8361	7	72.7025	3	47.5142	3	.284680	3	.08508	2	7.7809764	1
38130.C	32C.7137	4	67.5661	2	47.5245	1	.284635	3	.209422	2	7.7809848	4
38134.0	325.586	4	62.452	2	47.5233	3	.284639	7	.33325	1	7.780955	1
38138.C	33C.419	6	57.343	3	47.518	1	.28468	1	.45711	2	7.780971	2
38142.C	335.289	7	52.257	3	47.5215	8	.28479	1	.58076	2	7.780979	2
38146.C	34C.079	5	47.106	2	47.539	1	.28491	1	.70488	1	7.780993	4
38150.C	345.008	6	41.957	4	47.532	?	.28463	2	.82876	2	7.780960	1
38154.C	349.856	6	36.882	3	47.510	2	.28483	2	.95257	1	7.780972	3
38158.0	354.707	4	31.754	4	47.5275	4	.28474	2	.07637	1	7.780942	2
38162.C	359.563	6	26.655	2	47.5270	5	.28461	2	.20019	1	7.780964	3
38166.C	4.414	2	21.508	1	47.5257	6	.284773	6	.324159	5	7.781005	1
38170.C	9.270	3	16.38914	9	47.5258	3	.284847	2	.4479996	5	7.7809506	2
38174.0	14.131	2	11.2655	6	47.5212	2	.284795	2	.571811	3	7.7809576	4
38178.0	18.9624	6	6.1494	5	47.5197	2	.284857	1	.695673	1	7.7809559	2
38182.C	23.8214	5	1.0286	2	47.5214	2	.284913	3	.819471	2	7.7809479	2
38186.C	28.6762	6	355.90992	7	47.52187	6	.284939	2	.943277	2	7.7809532	3
38190.C	33.512	1	350.7959	5	47.5181	3	.284912	2	.067096	2	7.7809502	2
38194.C	38.348	1	345.6768	6	47.5163	5	.284946	3	.190924	2	7.7809432	3
38198.C	43.197	1	340.5565	7	47.5129	5	.284999	3	.314696	3	7.7809419	5
38202.0	48.028	5	335.4303	4	47.5101	3	.284945	2	.438516	1	7.7809404	4
38206.C	52.8686	5	330.3119	4	47.5123	2	.285030	2	.562292	1	7.7809360	3
38210.C	57.696	1	325.2005	1	47.5104	2	.285036	2	.686084	2	7.7809372	2

Table 11

RELATIVE POSITIONS OF THE SUN AND THE PERIGEE OF SATELLITE 1962 BETA UPSILON 1

MJD	Z	φ	ψ	D. R.A.	\dot{P}
PERIGEE IN SUNLIGHT					
38122.	1324.	-33.8	47.2	28.4	0.330E-07
38126.	1322.	-30.9	44.0	24.3	-0.760E-07
38130.	1322.	-27.8	40.6	19.8	0.793E-07
38134.	1321.	-24.6	37.1	15.1	0.162E-06
38138.	1319.	-21.4	33.7	10.1	-0.297E-06
38142.	1317.	-18.0	30.6	5.0	-0.496E-06
38146.	1315.	-14.6	28.1	359.6	-0.116E-05
38150.	1318.	-11.0	26.4	354.2	0.363E-06
38154.	1315.	-7.5	26.0	348.6	0.330E-06
38158.	1316.	-3.9	26.9	343.0	-0.991E-08
38162.	1317.	-0.3	29.0	337.2	0.661E-07
38166.	1316.	3.3	32.1	331.4	0.429E-06
38170.	1315.	6.8	35.8	325.6	-0.198E-07
38174.	1316.	10.4	39.9	319.9	0.231E-07
38178.	1316.	13.9	44.2	314.1	-0.661E-08
38182.	1316.	17.3	48.5	308.5	0.198E-07
38186.	1316.	20.7	52.8	302.9	0.155E-06
38190.	1318.	24.0	56.9	297.5	0.496E-07
38194.	1318.	27.2	60.7	292.2	0.760E-07
38198.	1319.	30.3	64.2	287.2	0.661E-08
38202.	1320.	33.2	67.4	282.5	0.297E-07
38206.	1320.	36.0	70.3	278.0	-0.165E-07
38210.	1321.	38.6	72.9	273.9	0.661E-07

The following satellites sometimes have a two-figure error in the column n'/2 of the mean elements:

1959	Eta
1959	Iota 1
1960	Xi 1

NOTICE

This series of Special Reports was instituted under the supervision of Dr. F. L. Whipple, Director of the Astrophysical Observatory of the Smithsonian Institution, shortly after the launching of the first artificial earth satellite on October 4, 1957. Contributions come from the Staff of the Observatory. First issued to ensure the immediate dissemination of data for satellite tracking, the Reports have continued to provide a rapid distribution of catalogues of satellite observations, orbital information, and preliminary results of data analyses prior to formal publication in the appropriate journals.

Edited and produced under the supervision of Mr. E. N. Hayes and Mrs. Barbara J. Mello, the reports are indexed by the Science and Technology Division of the Library of Congress, and are regularly distributed to all institutions participating in the U. S. space research program and to individual scientists who request them from the Administrative Officer, Technical Information, Smithsonian Astrophysical Observatory, Cambridge, Massachusetts 02138.